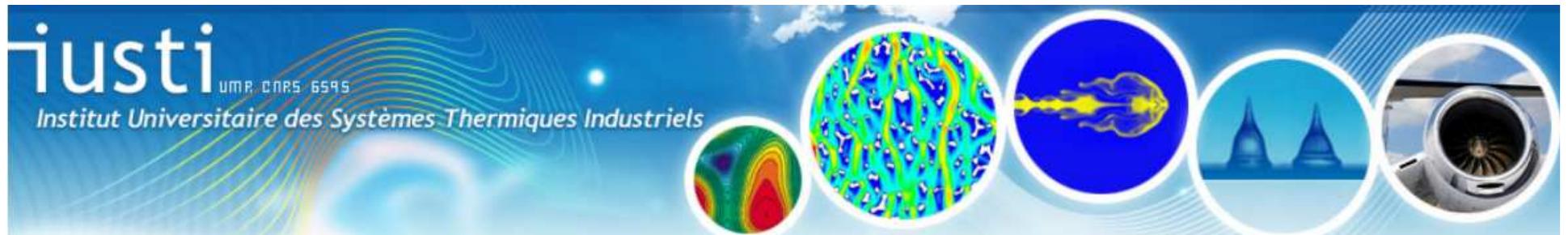
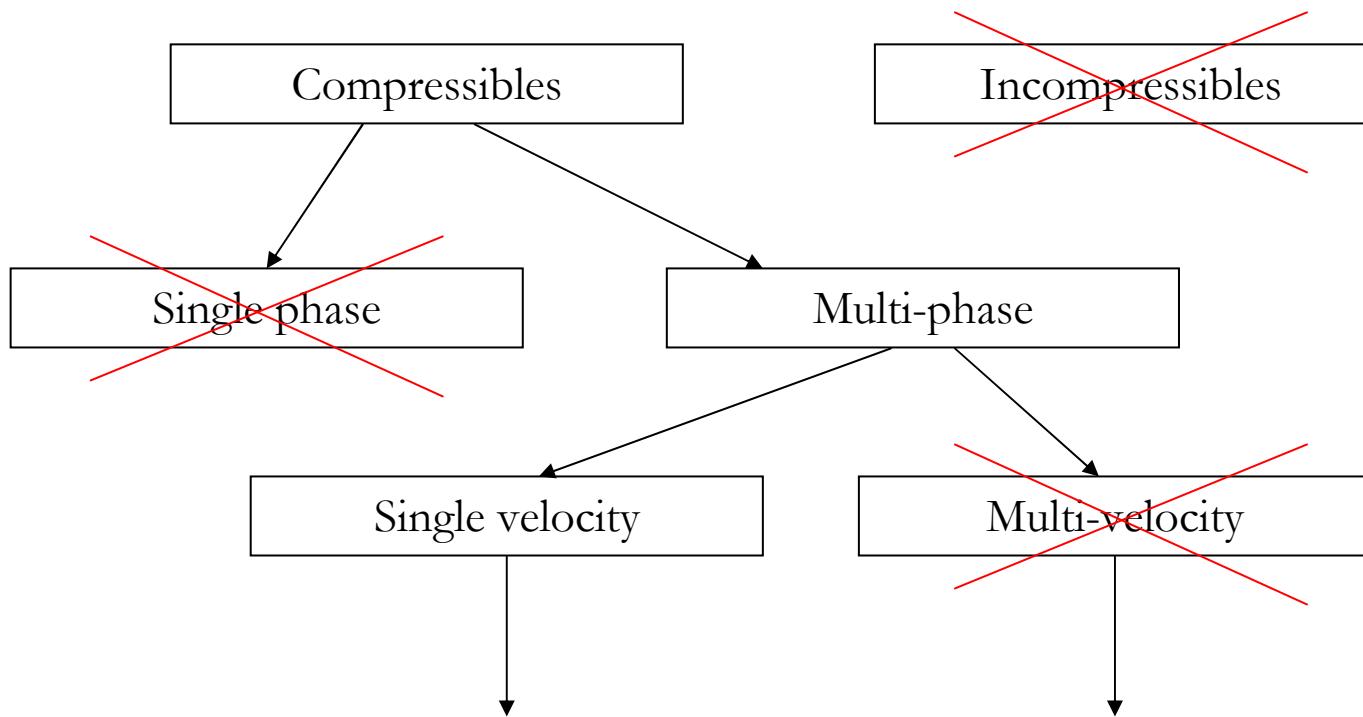


Numerical simulation of compressible two-phase flows

F. Petitpas
SMASH Team - MARSEILLE



Flows zoology – Position of the topic

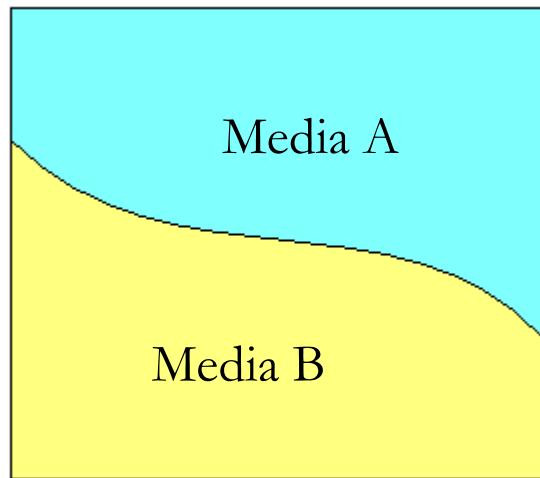


- Interface problems resolution
- Two-phase mixtures in mechanical equilibrium

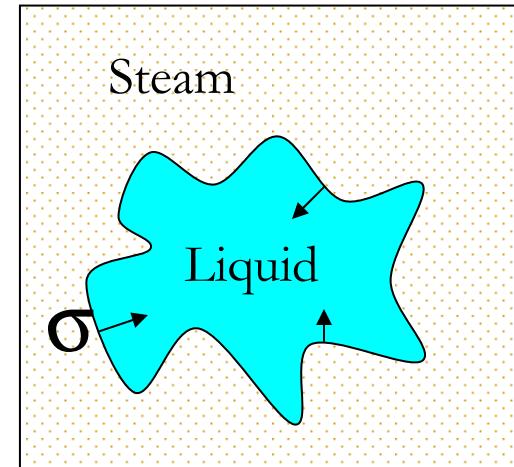
- Non equilibrium two-phase mixtures
- General models

Single velocity flows / Interface problems

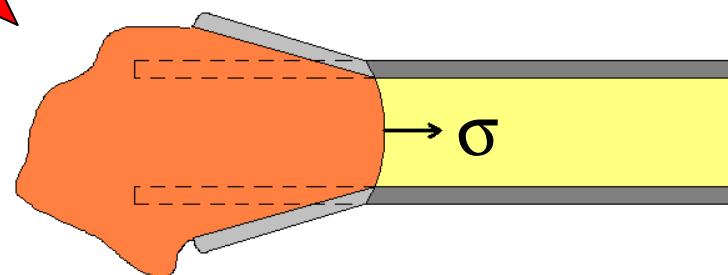
Single contact interface



Evaporation front

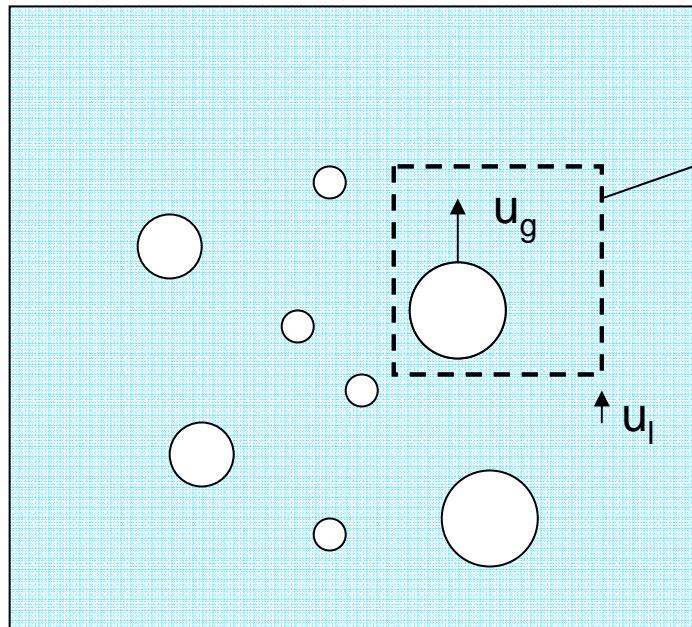


Detonation

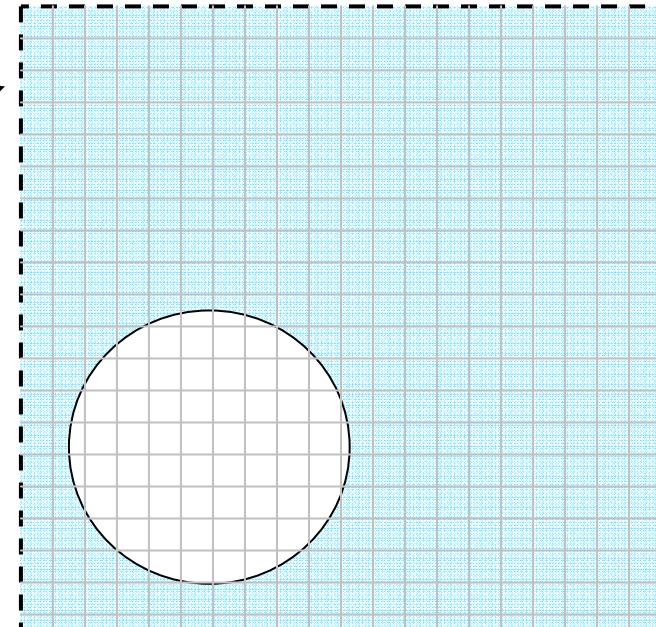


Some words about discretisation scales

“Macro” scale : Multi-velocity flow



« Micro » scale : Single velocity flow



An averaged model with two
velocities is needed

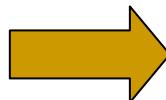
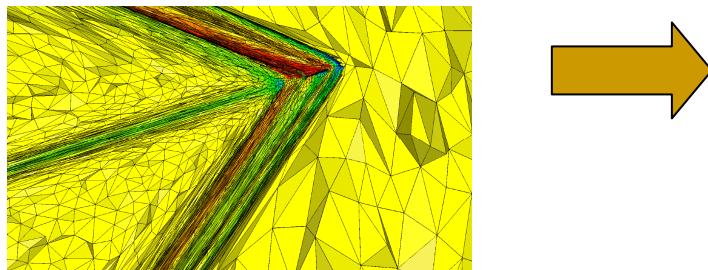


A reduced model for interface
problems is enough

On the choice of the method

■ Sharp interface methods

- ❑ Lagrangian methods with moving meshes, ALE (Arbitrary Lagrangian Eulerian)



Good for solid weak deformation



Not adapted for fluids computation with extreme deformations

- ❑ Front tracking, VOF, Level Set

 Very impressive results

Generally non conservative

 regarding mass and energy

Heavy numerical treatment

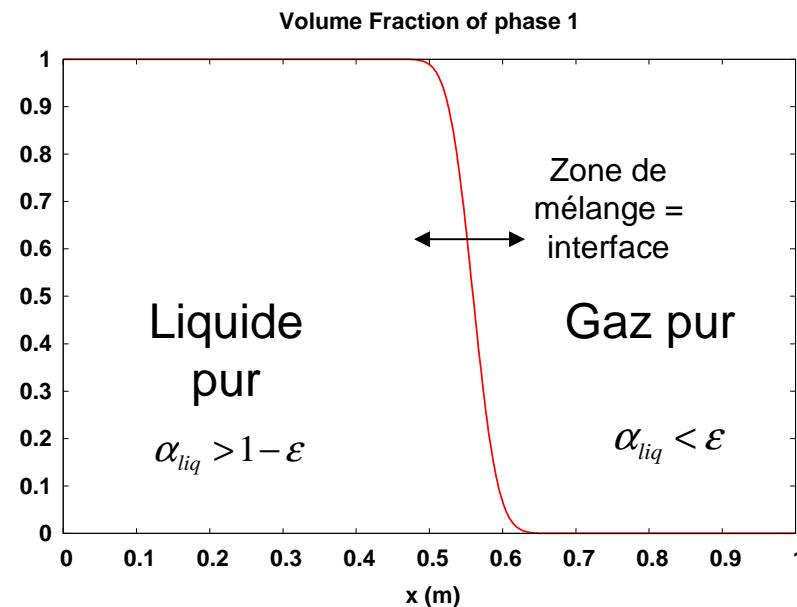
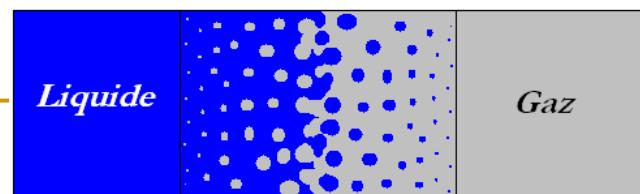
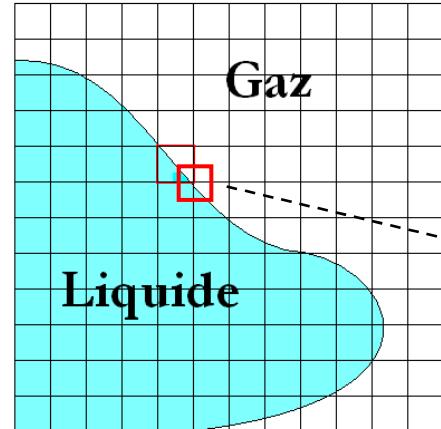


On the choice of the method

■ Diffuse interface methods

These methods authorize numerical diffusion of interfaces. This presents several advantages:

- Interfaces are not tracked or reconstructed, they are captured by the numerical scheme as artificial diffusion zone.
- By the way disappearance or apparition of interfaces are naturally obtained
- Conservative



On the choice of the model

- Euler equations with liquid-vapor Equation of state for evaporation problems
Single phase model with equilibrium EOS (T,p,g,u)
 - Able to compute liquid-vapor mixtures at Thermodynamical equilibrium
 - But metastable states are omitted
 - Unable to treat liquid-gas interfaces
- Multi-phase models
 - 4-equation : Euler + mass equation
Thermal and mechanical equilibrium (T,p,u)
 - Largely use for gas mixtures where thermal equilibrium condition is not so restrictive
 - But unable to treat simple contact interface (interface condition of equal pressure and velocity are violated)
 - 5-equation : 4 equations + volume fraction equation (Kapila et al., 2001)
Mechanical equilibrium (p,u)
 - Able to treat interfaces between non miscible fluids (liquid-gas)
 - Able to treat mixture evolving in mechanical equilibrium
 - 6-equation model
Velocity equilibrium (u)
 - 7-equation (Baer & Nunziato, 1986)
Total disequilibrium
 - Able to solve a large scale of problems
 - Difficult to solve numerically



Outline : The 5-equation model

- Topic 1 : Origins and properties of the 5-equation model
- Topic 2 : Numerical resolution
 - The Euler equations
 - The 5-equation model
- Topic 3 : Phase transition with the 5-equation model
- Topic 4 : Other extensions
 - Capillary effects
 - Compaction effects
 - Low Mach computing
 - Etc.



Topic 1 : Origins and properties of the 5-equation model

Starting point : origin of the 5-equation model (The 7-equation model)

Each phase obeys its own thermodynamics (pressure, density, internal energy) and has its own set of equations :

$$\frac{\partial \alpha_k}{\partial t} + \vec{\sigma} \cdot \vec{\nabla} \alpha_k = \mu (P_k - P_{k'})$$

$$\frac{\partial \alpha_k \rho_k}{\partial t} + \vec{\nabla} \cdot (\alpha_k \rho_k \vec{u}_k) = 0$$

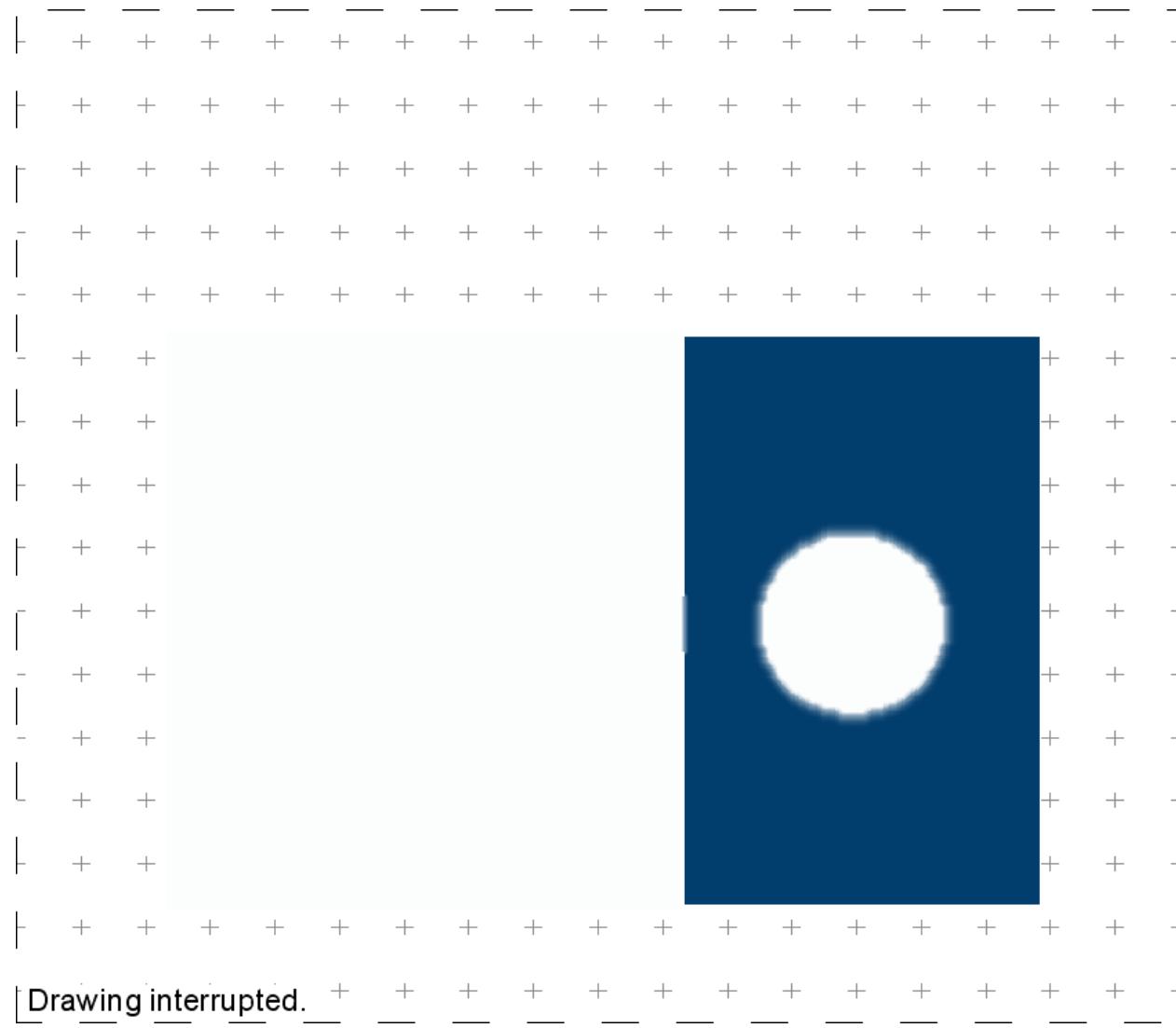
$$\frac{\partial \alpha_k \rho_k \vec{u}_k}{\partial t} + \vec{\nabla} \cdot (\alpha_k (\rho_k \vec{u}_k \otimes \vec{u}_k + P_k \vec{I})) = P_I \vec{\nabla} \alpha_k + \lambda (\vec{u}_{k'} - \vec{u}_k)$$

$$\frac{\partial \alpha_k \rho_k E_k}{\partial t} + \vec{\nabla} \cdot (\alpha_k (\rho_k E_k + P_k) \vec{u}_k) = P_I \vec{\sigma} \cdot \vec{\nabla} \alpha_k - \mu P_I (P_k - P_{k'}) + \lambda \vec{\sigma} \cdot (\vec{u}_{k'} - \vec{u}_k)$$

$$\vec{\sigma} = \vec{u}_2 \quad \text{and} \quad P_I = P_1 \qquad \qquad \text{Baer \& Nunziato (1986)}$$

$$\vec{\sigma} = \frac{Z_1 \vec{u}_1 + Z_2 \vec{u}_2}{Z_1 + Z_2} \quad \text{and} \quad P_I = \frac{Z_1 P_2 + Z_2 P_1}{Z_1 + Z_2} \quad \left\{ \begin{array}{l} \text{Saurel \& al. (2003)} \\ \text{Chinnayya \& al (2004)} \end{array} \right.$$

Example of interface problem evolving to a two-velocity mixture



Asymptotic reduction of the 7-equation model

- Why it is interesting to use the 5-equation model instead of the 7-equation one ?
 - It is difficult to solve and implies heavy costs regarding CPU and memory.
 - It contains extra unuseful physics to treat interface problems (two velocities and two pressures)
 - Extra physical effects are difficult to introduce (as for example phase transition, capillary effects, etc.)

Asymptotic reduction by the Chapman-Enskog method :

$$\lambda, \mu = 1/\epsilon \rightarrow \infty \quad \text{Relaxation parameters tend to infinity}$$

$$f = f^0 + \epsilon f^1 \quad \text{Each flow variable is supporting small variation around an equilibrium state}$$

The diffuse interface model (5-equation model)

4 conservative equations

$$\left\{ \begin{array}{l} \frac{d\alpha_1}{dt} = \alpha_1 \alpha_2 \left(\frac{\rho_2 c_2^2 - \rho_1 c_1^2}{\alpha_2 \rho_1 c_1^2 + \alpha_1 \rho_2 c_2^2} \right) \frac{\partial u}{\partial x} \\ \frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} = 0 \\ \frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} = 0 \\ \frac{\partial \rho u}{\partial t} + \frac{\partial (\rho u^2 + p)}{\partial x} = 0 \\ \frac{\partial \rho E}{\partial t} + \frac{\partial (\rho E + p)u}{\partial x} = 0 \end{array} \right.$$

Equilibre mécanique

$$u_1 = u_2 = u$$

$$p_1 = p_2 = p$$

Variables de mélange :

$$\rho = \alpha_1 \rho_1 + \alpha_2 \rho_2$$

$$\rho E = \alpha_1 \rho_1 E_1 + \alpha_2 \rho_2 E_2$$

+ Équation d'état de mélange :

$$p = p(\rho, e, \alpha_k) = \frac{\rho e - \left(\frac{\alpha_1 \gamma_1 p_{\infty,1}}{\gamma_1 - 1} + \frac{\alpha_2 \gamma_2 p_{\infty,2}}{\gamma_2 - 1} \right)}{\frac{\alpha_1}{\gamma_1 - 1} + \frac{\alpha_2}{\gamma_2 - 1}}$$

We will come back on thermodynamic closure in the following ...



This is a mechanical equilibrium but each phase remains in thermal disequilibrium 13

Physical meaning of the volume fraction equation

$\rho_k c_k^2$ is the Bulk modulus of media k

It traduces the compressibility of a media

- Big when it is weakly compressible

- Small when it is strongly compressible

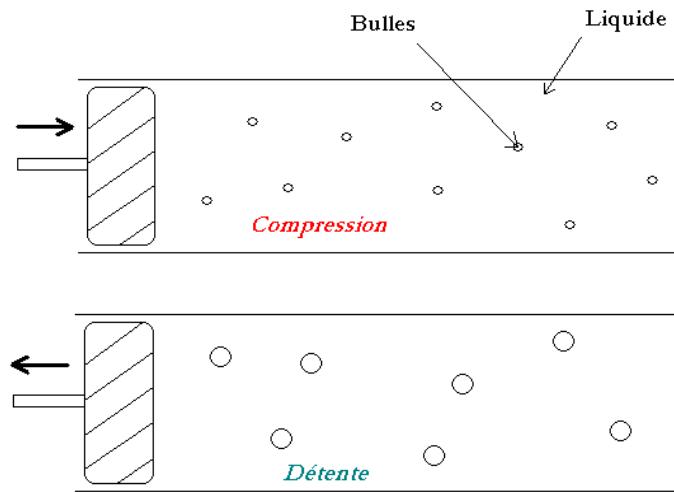
$$\begin{cases} \frac{d\alpha_1}{dt} = \alpha_1 \alpha_2 \left(\frac{\rho_2 c_2^2 - \rho_1 c_1^2}{\alpha_2 \rho_1 c_1^2 + \alpha_1 \rho_2 c_2^2} \right) \frac{\partial u}{\partial x} \\ \frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} = 0 \\ \frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} = 0 \\ \frac{\partial \rho u}{\partial t} + \frac{\partial (\rho u^2 + p)}{\partial x} = 0 \\ \frac{\partial \rho E}{\partial t} + \frac{\partial (\rho E + p)u}{\partial x} = 0 \end{cases}$$

$$\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} = K \frac{\partial u}{\partial x}$$

Exemple $K < 0$

Liquide : phase 1

Bulles gaz : phase 2

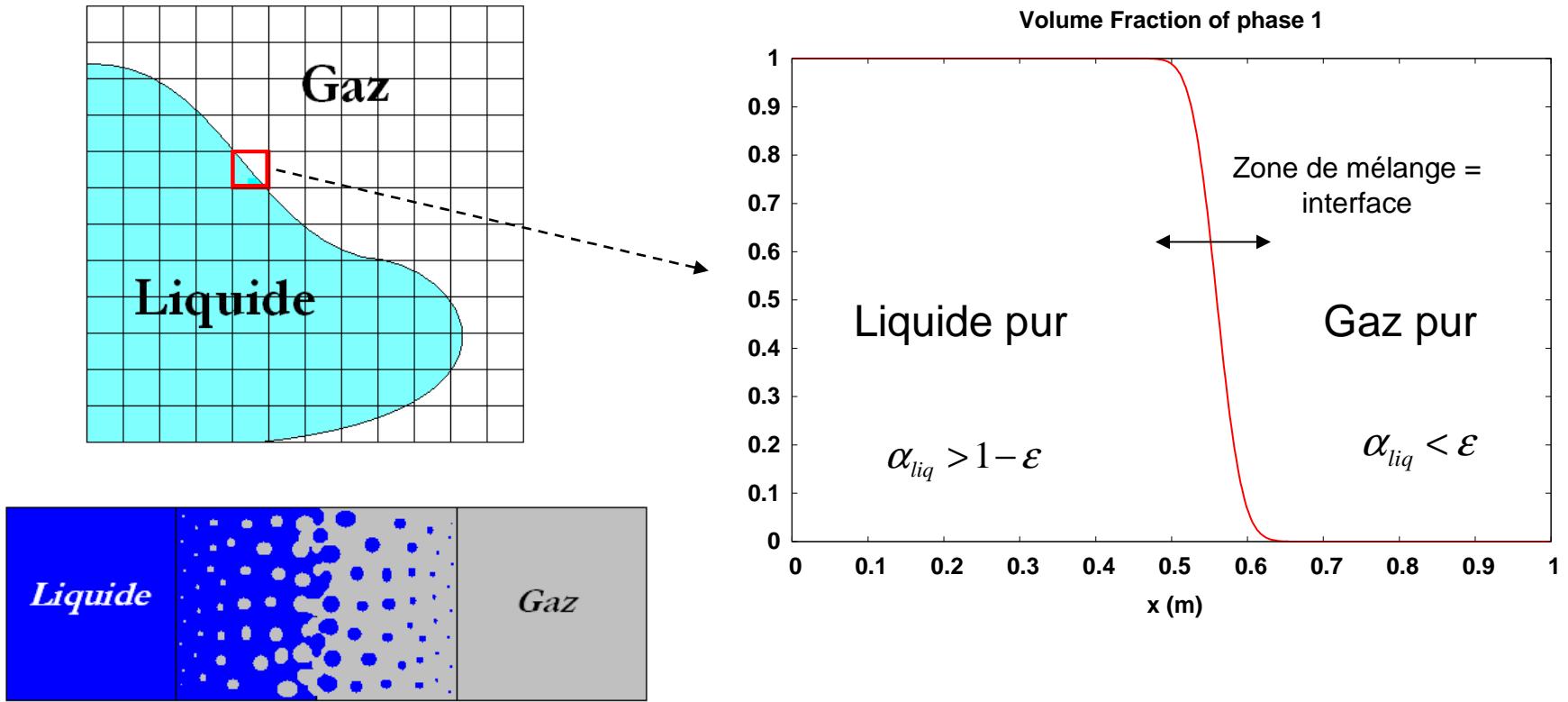


$\frac{\partial u}{\partial x} < 0 \Rightarrow \alpha_1$ increases

$\frac{\partial u}{\partial x} > 0 \Rightarrow \alpha_1$ decreases

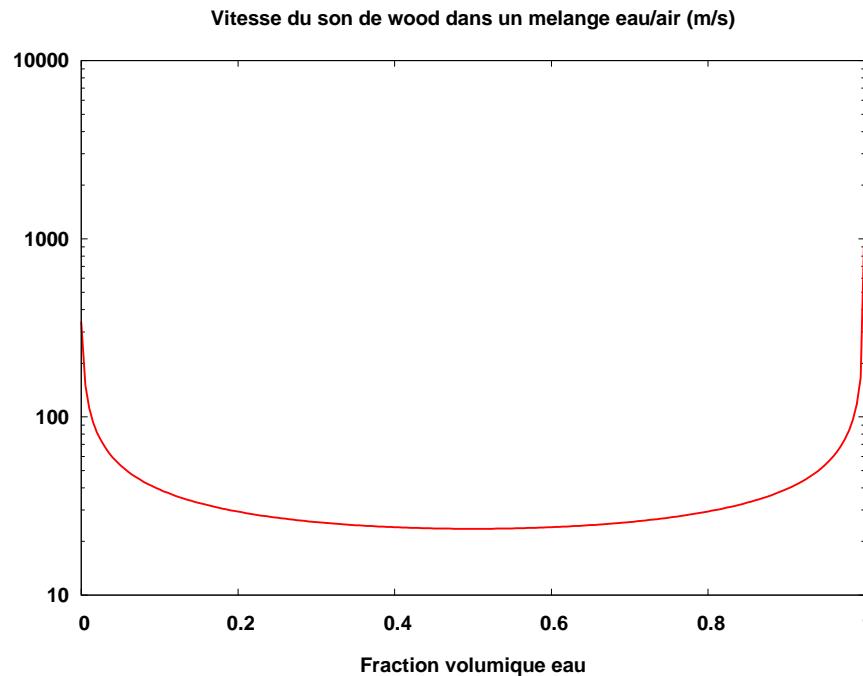
Back to diffuse interface aptitude ...

- Diffusion is due to numerical treatment : Artificial diffusion
- By the way, an interface looks like a mixture zone : $\varepsilon < \alpha_k < 1 - \varepsilon$
- The interface conditions of pressures and velocities equalities are automatically obtained !



5-equation model properties

- Hyperbolic systems : 3 waves speeds.
 $u - c_w, u, u + c_w$
- The speed of sound is those of Wood (1930)



$$\boxed{\begin{aligned} \frac{d\alpha_1}{dt} &= K \operatorname{div}(\vec{u}) \\ \frac{\partial \alpha_1 \rho_1}{\partial t} + \operatorname{div}(\alpha_1 \rho_1 \vec{u}) &= 0 \\ \frac{\partial \alpha_2 \rho_2}{\partial t} + \operatorname{div}(\alpha_2 \rho_2 \vec{u}) &= 0 \\ \frac{\partial \rho u}{\partial t} + \operatorname{div}(\rho \vec{u} \otimes \vec{u}) + \vec{\nabla} p &= 0 \\ \frac{\partial \rho E}{\partial t} + \operatorname{div}[(\rho E + p) \vec{u}] &= 0 \end{aligned}}$$

$$\frac{1}{\rho c_w^2} = \frac{\alpha_1}{\rho_1 c_1^2} + \frac{\alpha_2}{\rho_2 c_2^2}$$

Topic 2 : Numerical resolution considerations

- Basics for Euler conservative equations
- 5-equation model numerical resolution

Basics of numerical resolution for the Euler equations

- Euler equations:

- Mass balance

$$\frac{\partial \rho}{\partial t} + \operatorname{div}(\rho \vec{u}) = 0$$

- Momentum balance

$$\frac{\partial \rho \vec{u}}{\partial t} + \operatorname{div}(\rho \vec{u} \otimes \vec{u}) + \operatorname{grad}(P) = 0$$

- Total energy balance

$$\frac{\partial \rho E}{\partial t} + \operatorname{div}((\rho E + P) \vec{u}) = 0$$

- 1D simplification:

$$\frac{\partial \rho}{\partial t} + \frac{\partial \rho u}{\partial x} = 0$$

$$\frac{\partial \rho u}{\partial t} + \frac{\partial \rho u^2 + P}{\partial x} = 0$$

$$\frac{\partial \rho E}{\partial t} + \frac{\partial u(\rho E + P)}{\partial x} = 0$$

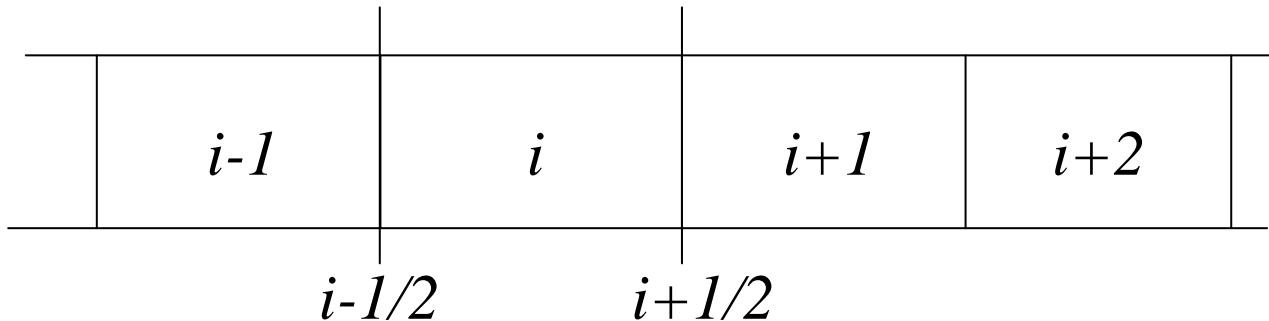


$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial x} = 0$$

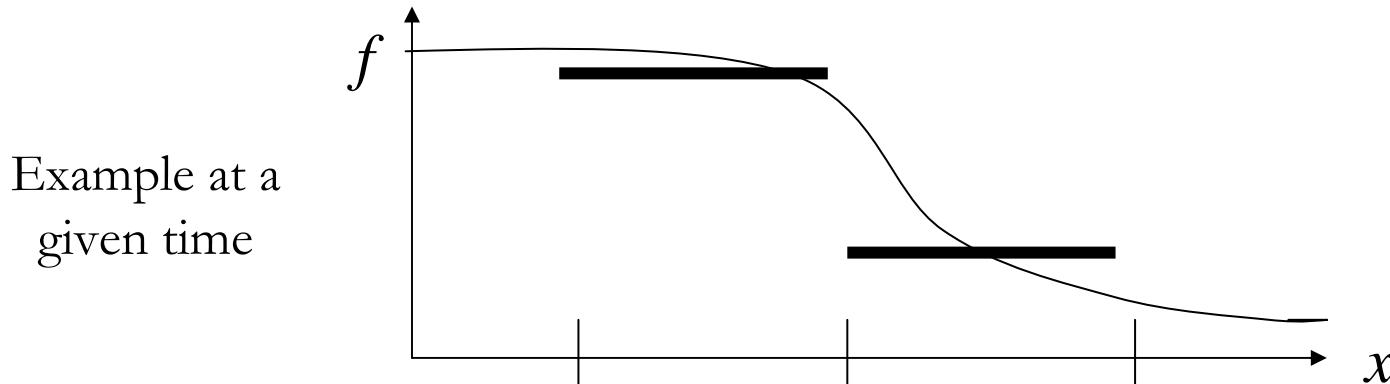
$$U = (\rho, \rho u, \rho E)^T$$

$$F = (\rho u, \rho u^2 + P, u(\rho E + P))^T$$

Computational mesh

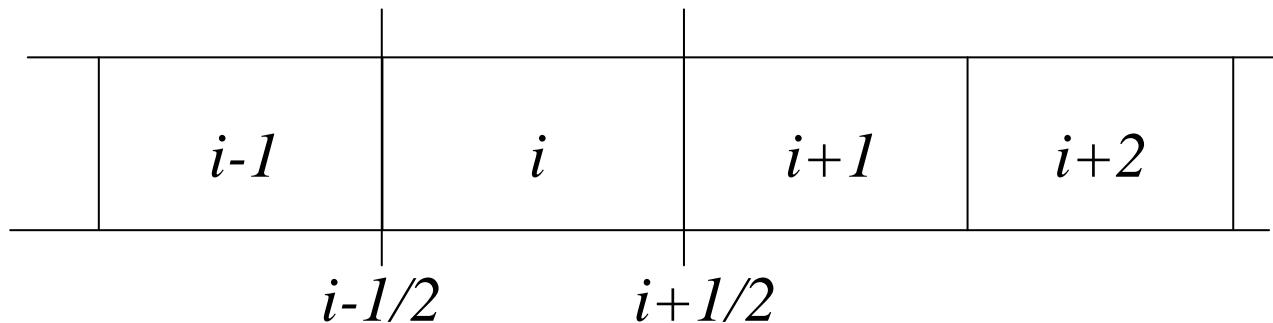


- The cell ‘ i ’ is bounded by inlet and outlet sections $i-1/2$ and $i+1/2$.
- The fluxes cross over these cell boundaries.
- The unknowns are computed at the cell center and are piecewise constant functions in the cell.



How to obtain average cells variables form one time step to another one

Numerical approximation



Integration on cell i over time :

$$\iint_{\Delta t V} \left[\frac{\partial U}{\partial t} + \operatorname{div}(F) \right] dV dt = 0 \quad \longrightarrow \quad U_i^{n+1} = U_i^n - \frac{\Delta t}{\Delta x} (F_{i+1/2}^* - F_{i-1/2}^*)$$

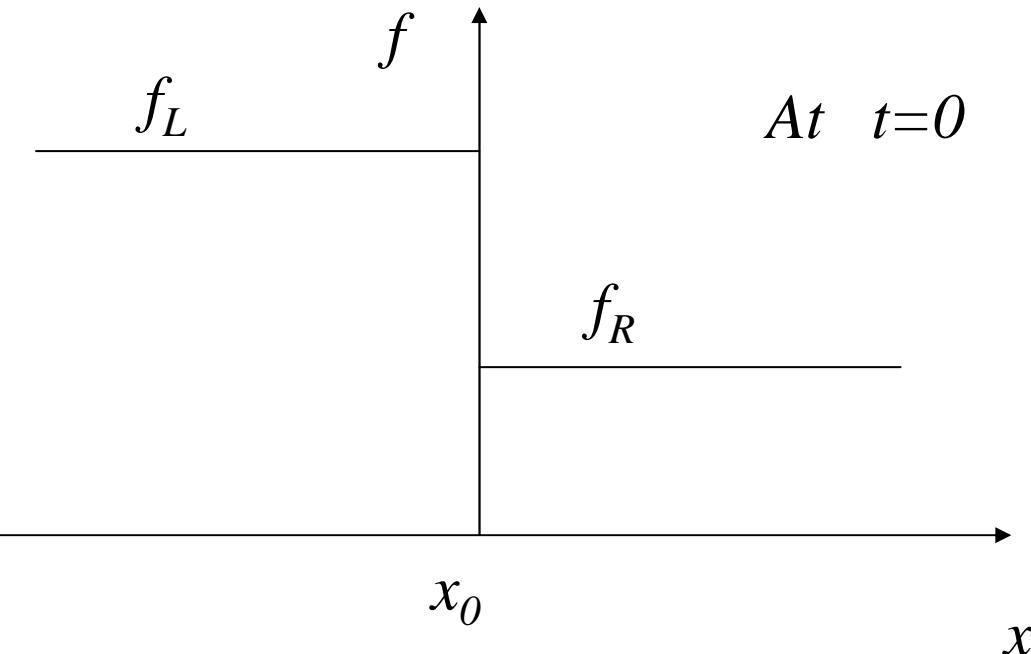
The '*' superscript is for : Solution of the Riemann problem at cell boundary



The Flux F at cell boundaries has been supposed to be constant during integration time step

→ CFL condition

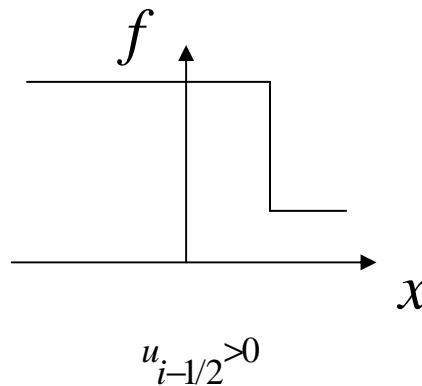
The Riemann problem



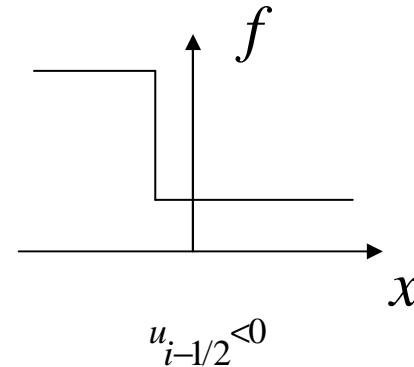
- The initial data are known at a given time and are constant on the right and left side.
- A discontinuity connects the two states.
- Question: How does the solution evolve at $t>0$?

The Riemann problem solution (advection)

The two possible solutions are:



$$u_{i-1/2} > 0$$

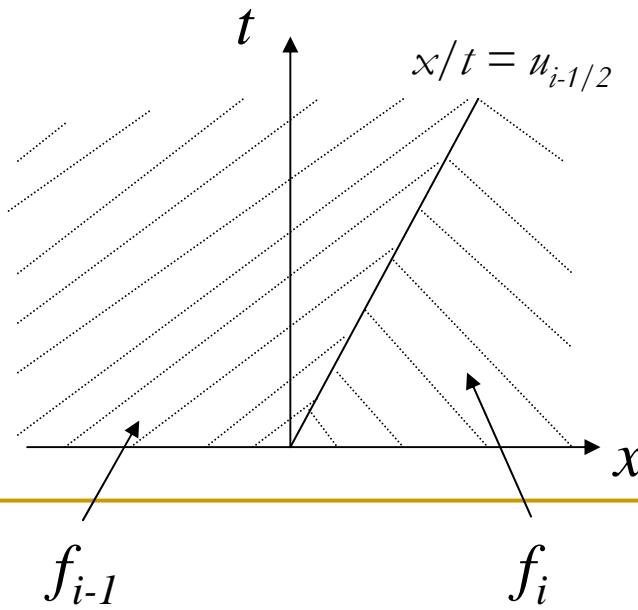


$$u_{i-1/2} < 0$$

In the (x,t) diagram

The solution is:

$$f(x/t) = \begin{cases} f_{i-1} & \text{if } x/t < u_{i-1/2} \\ f_i & \text{if } x/t > u_{i-1/2} \end{cases}$$



For the linearized Euler equations

$$\frac{\partial \rho}{\partial t} + \frac{\partial \rho u}{\partial x} = 0$$

$$\frac{\partial \rho u}{\partial t} + \frac{\partial \rho u^2 + P}{\partial x} = 0$$

$$\frac{\partial \rho E}{\partial t} + \frac{\partial u(\rho E + P)}{\partial x} = 0$$



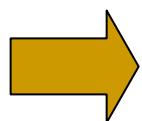
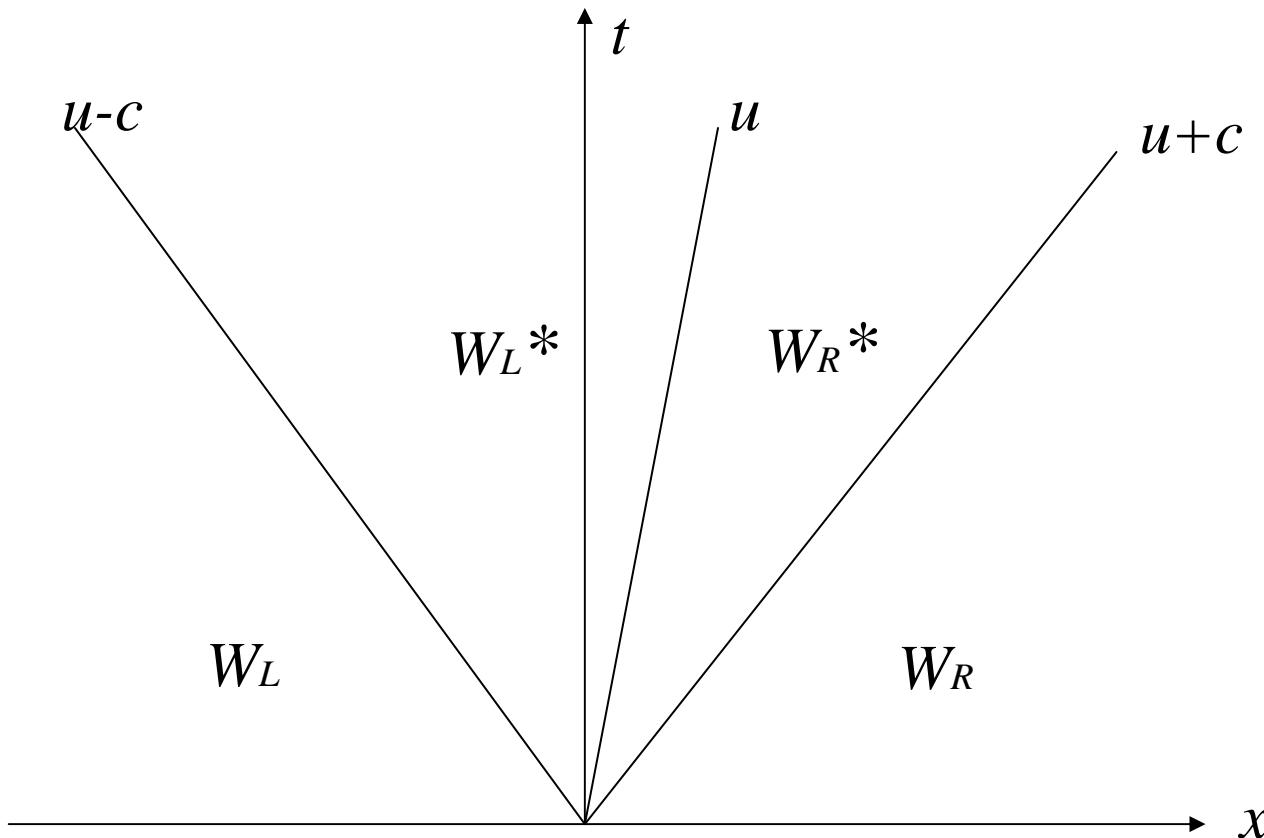
$$\left\{ \begin{array}{l} \frac{\partial W}{\partial t} + A(W) \frac{\partial W}{\partial x} = 0 \quad \text{analogue of } \frac{\partial f}{\partial t} + u \frac{\partial f}{\partial x} = 0 \\ W = (\rho, u, P)^T \\ A(W) = \begin{pmatrix} u & \rho & 0 \\ 0 & u & 1/\rho \\ 0 & \rho c^2 & u \end{pmatrix} \end{array} \right.$$

Plays the role of a propagation velocity

The eigenvalues of A are the waves speeds:

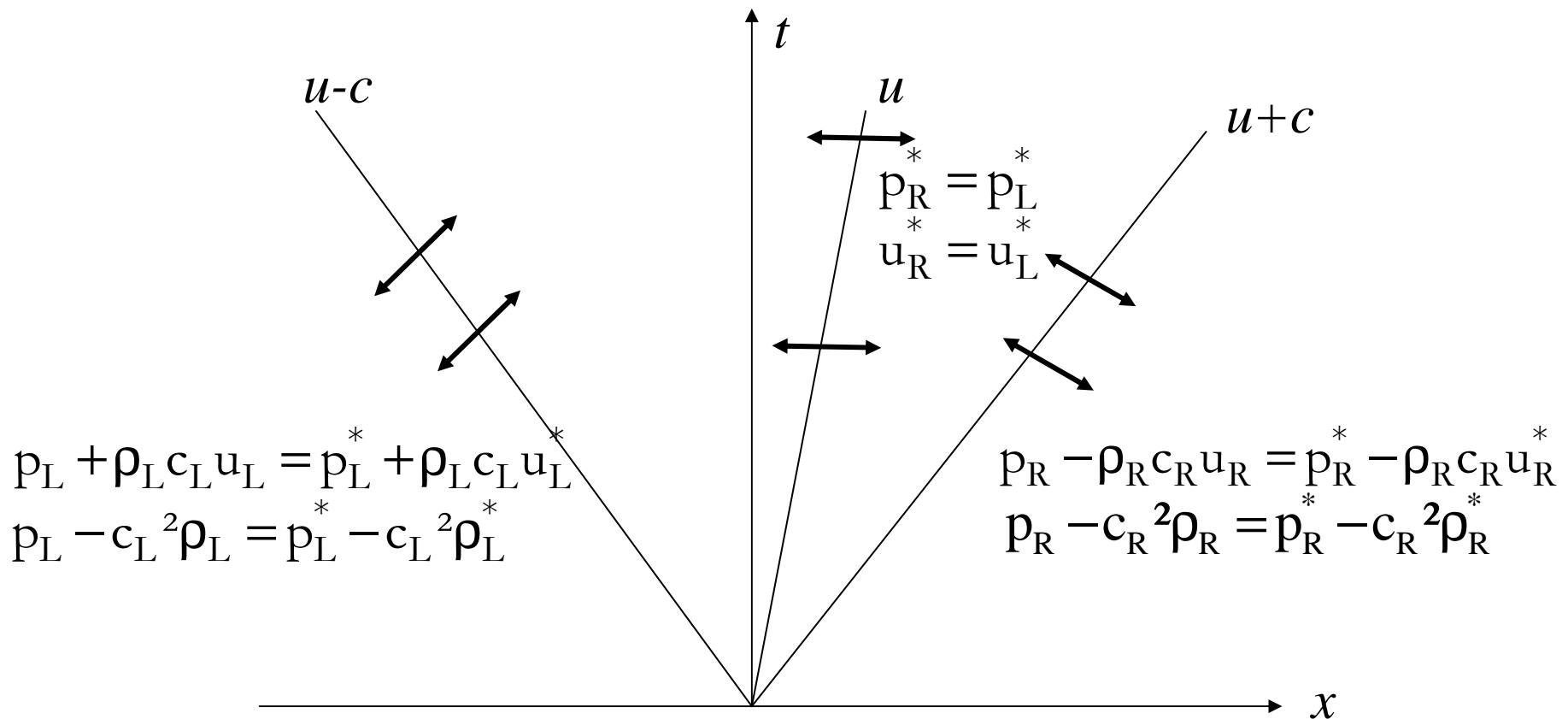
$$\lambda^+ = u + c, \quad \lambda^- = u - c, \quad \lambda^0 = u$$

The Riemann problem for linearized Euler equations



Solving the Riemann problem consist in determining the perturbed states W_L^* and W_R^* after waves propagation from the known states W_L and W_R .

The Riemann problem solution for linearized Euler equations



From this algebraic set of 6 equations the two intermediate states

W_L^* and W_R^* are readily obtained.

Riemann problem solution

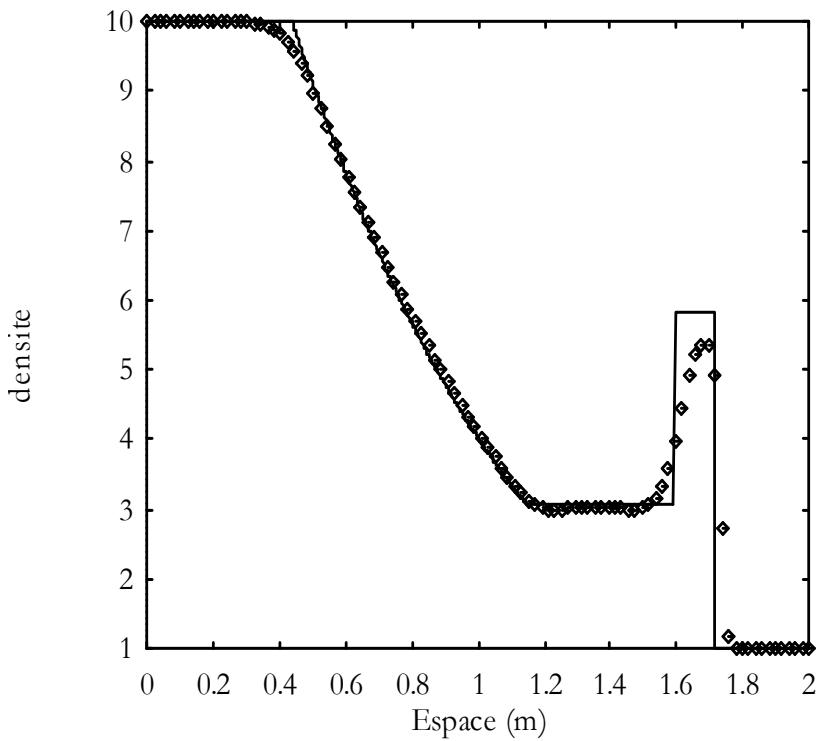
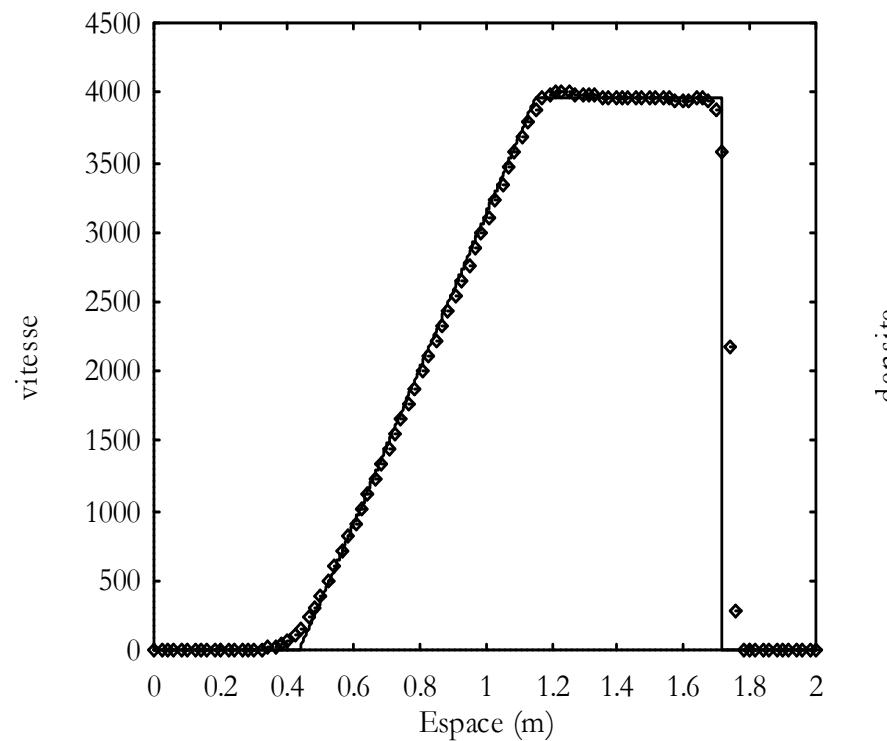
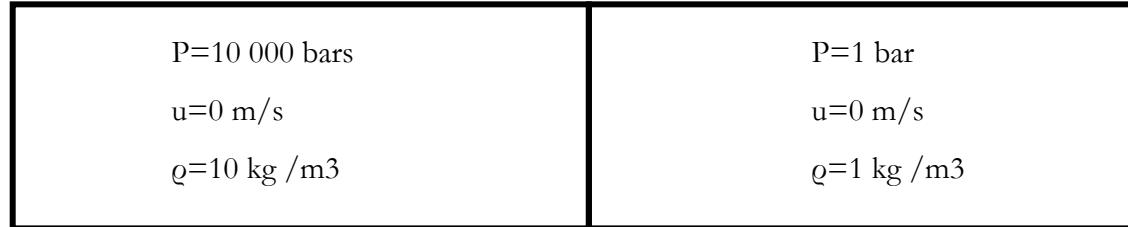
$$u^* = \frac{p_L - p_R + Z_R u_R + Z_L u_L}{Z_R + Z_L}$$

$$p^* = \frac{Z_R p_L + Z_L p_R + Z_R Z_L (u_L - u_R)}{Z_R + Z_L}$$

$$\rho_R^* = \rho_R + \frac{p^* - p_R}{c_R^2}$$

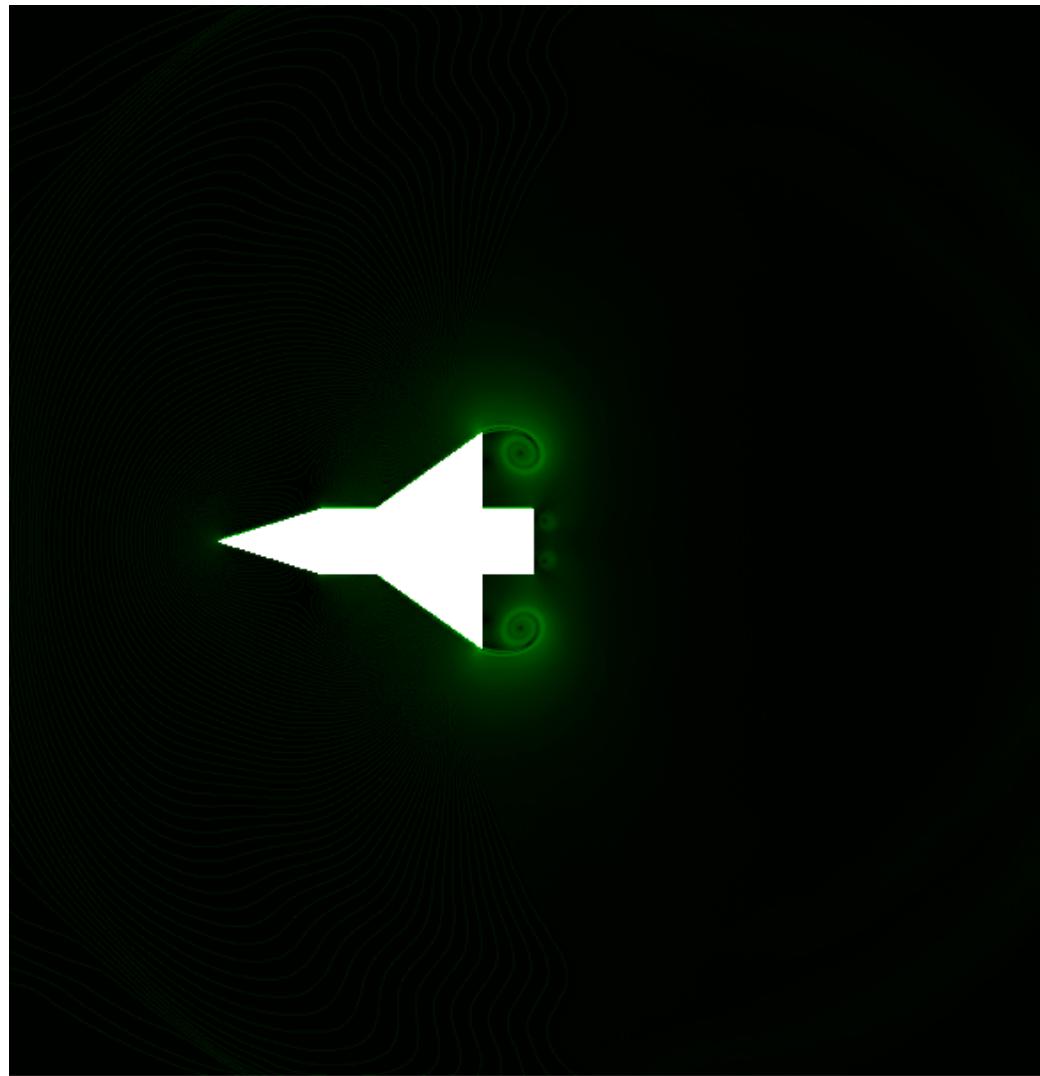
$$\rho_L^* = \rho_L + \frac{p^* - p_L}{c_L^2}$$

Example: shock tube



Exact solution (lines) / Computed results with Godunov (symbols)

Example – Supersonic flow around plane profil



Summary for Euler equations

- The Riemann problem solution is a local solution of the Euler equations between two discontinuous initial states.
- It is the cornerstone of all numerical schemes used in gas dynamics, shallow water and modern multiphase codes

- Recommended literature:
E.F. Toro (1997) Riemann solvers and numerical methods for fluid dynamics.
Springer Verlag

The diffuse interface model (5-equation model)

4 conservative equations

$$\left\{ \begin{array}{l} \frac{d\alpha_1}{dt} = \alpha_1 \alpha_2 \left(\frac{\rho_2 c_2^2 - \rho_1 c_1^2}{\alpha_2 \rho_1 c_1^2 + \alpha_1 \rho_2 c_2^2} \right) \frac{\partial u}{\partial x} \\ \frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} = 0 \\ \frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} = 0 \\ \frac{\partial \rho u}{\partial t} + \frac{\partial (\rho u^2 + p)}{\partial x} = 0 \\ \frac{\partial \rho E}{\partial t} + \frac{\partial (\rho E + p)u}{\partial x} = 0 \end{array} \right.$$

Equilibre mécanique

$$u_1 = u_2 = u$$

$$p_1 = p_2 = p$$

Variables de mélange :

$$\rho = \alpha_1 \rho_1 + \alpha_2 \rho_2$$

$$\rho E = \alpha_1 \rho_1 E_1 + \alpha_2 \rho_2 E_2$$

+ Équation d'état de mélange :

$$p = p(\rho, e, \alpha_k) = \frac{\rho e - \left(\frac{\alpha_1 \gamma_1 p_{\infty,1}}{\gamma_1 - 1} + \frac{\alpha_2 \gamma_2 p_{\infty,2}}{\gamma_2 - 1} \right)}{\frac{\alpha_1}{\gamma_1 - 1} + \frac{\alpha_2}{\gamma_2 - 1}}$$

We will come back on thermodynamic closure in the following ...



This is a mechanical equilibrium but each phase remains in thermal disequilibrium 30

Numerical resolutions: issues

- 1) Volume fraction positivity: How to treat the non-conservative term in the volume fraction equation when shocks or strong rarefaction waves are present ?

$$\frac{\partial \alpha_1}{\partial t} + \vec{u} \cdot \vec{\nabla} \alpha_1 = \frac{(\rho_2 c_2^2 - \rho_1 c_1^2)}{\frac{\rho_1 c_1^2}{\alpha_1} + \frac{\rho_2 c_2^2}{\alpha_2}} \vec{\nabla} \cdot \vec{u}$$

Difficulty to guarantee that $0 < \alpha_1 < 1$

- 2) The volume fraction varies across acoustic waves: Riemann solver difficult to construct.

6+1-equation model

- Previous difficulties are circumvented using a pressure non-equilibrium model

$$\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} = \mu(p_1 - p_2)$$

$$\frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} = 0$$

$$\frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} = 0$$

$$\frac{\partial \rho u}{\partial t} + \frac{\partial \rho u^2 + (\alpha_1 p_1 + \alpha_2 p_2)}{\partial x} = 0$$

$$\frac{\partial \alpha_1 \rho_1 e_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 e_1 u}{\partial t} + \alpha_1 p_1 \frac{\partial u}{\partial x} = -p_I \mu (p_1 - p_2)$$

$$\frac{\partial \alpha_2 \rho_2 e_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 e_2 u}{\partial t} + \alpha_2 p_2 \frac{\partial u}{\partial x} = p_I \mu (p_1 - p_2)$$

6 equations + 1 redundant equation (coming from the summation of energies):

$$\rho = \alpha_1 \rho_1 + \alpha_2 \rho_2$$

$$\rho E = \alpha_1 \rho_1 E_1 + \alpha_2 \rho_2 E_2$$

- The pressure equilibrium 5-equation model is obtained from this 6-equation model in the asymptotic limit of stiff pressure relaxation coefficient,

- The speed of sound is monotonic,

$$c_f^2 = Y_1 c_1^2 + Y_2 c_2^2$$

- The volume fraction is constant through right- and left-facing waves when relaxation effects are absent ($\mu=0$).

$$\frac{\partial \rho(Y_1 e_1 + Y_2 e_2 + \frac{1}{2} u^2)}{\partial t} + \frac{\partial u \left(\rho(Y_1 e_1 + Y_2 e_2 + \frac{1}{2} u^2) + (\alpha_1 p_1 + \alpha_2 p_2) \right)}{\partial x} = 0$$

3-step methods

- a) The (6+1)-equation model is solved without relaxation effects: Godunov-type scheme,
- b) Stiff pressure relaxation procedure,
- c) Energies reset (in order to ensure energy conservation)



The 5-equation model is solved

1st step: Godunov-type scheme

- Without relaxation terms, the 6+1 equation model becomes:

$$\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} = 0$$

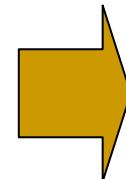
$$\frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} = 0$$

$$\frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} = 0$$

$$\frac{\partial \rho u}{\partial t} + \frac{\partial \rho u^2 + (\alpha_1 p_1 + \alpha_2 p_2)}{\partial x} = 0$$

$$\frac{\partial \alpha_1 \rho_1 e_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 e_1 u}{\partial t} + \alpha_1 p_1 \frac{\partial u}{\partial x} = 0$$

$$\frac{\partial \alpha_2 \rho_2 e_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 e_2 u}{\partial t} + \alpha_2 p_2 \frac{\partial u}{\partial x} = 0$$



An advection equation

$$\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} = 0$$

+ A conservative part

$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial x} = 0$$

with $U = ((\alpha \rho)_1, (\alpha \rho)_2, \rho u, \rho E)^T$

$$F = ((\alpha \rho)_1 u, (\alpha \rho)_2 u, \rho u^2 + p, (\rho E + p)u)^T$$

+ A non conservative one

$$\frac{\partial \alpha_1 \rho_1 e_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 e_1 u}{\partial t} + \alpha_1 p_1 \frac{\partial u}{\partial x} = 0$$

$$\frac{\partial \alpha_2 \rho_2 e_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 e_2 u}{\partial t} + \alpha_2 p_2 \frac{\partial u}{\partial x} = 0$$

1st step: Godunov-type scheme

Godunov scheme for advection equation

$$\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} = 0 \quad \longrightarrow \quad \alpha_{1i}^{n+1} = \alpha_{1i}^n - \frac{\Delta t}{\Delta x} \left((u \alpha_1)_{i+1/2}^* - (u \alpha_1)_{i-1/2}^* - \alpha_{1i}^n (u_{i+1/2}^* - u_{i-1/2}^*) \right)$$

+ Godunov scheme for conservative equations

$$\frac{\partial U}{\partial t} + \frac{\partial F}{\partial x} = 0 \quad \longrightarrow \quad U_i^{n+1} = U_i^n - \frac{\Delta t}{\Delta x} \left(F^*(U_i^n, U_{i+1}^n) - F^*(U_{i-1}^n, U_i^n) \right)$$

$$U = ((\alpha \rho)_1, (\alpha \rho)_2, \rho u, \rho E)^T$$

with

$$F = ((\alpha \rho)_1 u, (\alpha \rho)_2 u, \rho u^2 + p, (\rho E + p) u)^T$$

+ A non conventional scheme for non conservative internal energies equations

$$\begin{aligned} \frac{\partial \alpha_1 \rho_1 e_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 e_1 u}{\partial t} + \alpha_1 p_1 \frac{\partial u}{\partial x} &= 0 & (\alpha \rho e)_{ki}^{n+1} &= (\alpha \rho e)_{ki}^n \\ \frac{\partial \alpha_2 \rho_2 e_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 e_2 u}{\partial t} + \alpha_2 p_2 \frac{\partial u}{\partial x} &= 0 & -\frac{\Delta t}{\Delta x} \left((\alpha \rho e u)_{ki+1/2}^* - (\alpha \rho e u)_{ki-1/2}^* + (\alpha p)_{ki} (u_{i+1/2}^* - u_{i-1/2}^*) \right) \end{aligned} \quad \longrightarrow$$

Riemann solver

$$\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} = 0$$

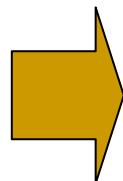
$$\frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} = 0$$

$$\frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} = 0$$

$$\frac{\partial \rho u}{\partial t} + \frac{\partial \rho u^2 + (\alpha_1 p_1 + \alpha_2 p_2)}{\partial x} = 0$$

$$\frac{\partial \alpha_1 \rho_1 e_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 e_1 u}{\partial t} + \alpha_1 p_1 \frac{\partial u}{\partial x} = 0$$

$$\frac{\partial \alpha_2 \rho_2 e_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 e_2 u}{\partial t} + \alpha_2 p_2 \frac{\partial u}{\partial x} = 0$$



$$\frac{\partial W}{\partial t} + A(W) \frac{\partial W}{\partial x} = 0$$

$$W = (\alpha_1, s_1, s_2, u, p_1, p_2)^T$$

$$A(W) = \begin{pmatrix} u & 0 & 0 & 0 & 0 & 0 \\ 0 & u & 0 & 0 & 0 & 0 \\ 0 & 0 & u & 0 & 0 & 0 \\ \frac{p_1 - p_2}{\rho} & 0 & 0 & u & \frac{\alpha_1}{\rho} & \frac{\alpha_2}{\rho} \\ 0 & 0 & 0 & \rho_1 c_1^2 & u & 0 \\ 0 & 0 & 0 & \rho_2 c_2^2 & 0 & u \end{pmatrix}$$

3 waves speeds $\lambda^+ = u + c_f$, $\lambda^- = u - c_f$, $\lambda^0 = u$

with $c_f^2 = Y_1 c_1^2 + Y_2 c_2^2$

Riemann problem solution

$$u^* = \frac{p_L - p_R + Z_R u_R + Z_L u_L}{Z_R + Z_L}$$

$$p^* = \frac{Z_R p_L + Z_L p_R + Z_R Z_L (u_L - u_R)}{Z_R + Z_L}$$

$$\alpha_{kL}^* = \alpha_{kL}$$

$$\alpha_{kR}^* = \alpha_{kR}$$

$$s_{kR}^* = s_{kR}$$

$$s_{kL}^* = s_{kL}$$

with

$$p = \alpha_1 p_1 + \alpha_2 p_2$$

$$\rho = \alpha_1 \rho_1 + \alpha_2 \rho_2$$

$$Z = \rho c$$

$$c_f^2 = Y_1 c_1^2 + Y_2 c_2^2$$



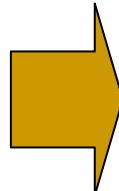
Similar to Euler Riemann solver

2nd step: Pressure relaxation

Already solved by the 1st step

$$\begin{aligned}\frac{\partial \alpha_1}{\partial t} + u \frac{\partial \alpha_1}{\partial x} &= \\ \frac{\partial \alpha_1 \rho_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 u}{\partial x} &= 0 \\ \frac{\partial \alpha_2 \rho_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 u}{\partial x} &= 0 \\ \frac{\partial \rho u}{\partial t} + \frac{\partial \rho u^2 + (\alpha_1 p_1 + \alpha_2 p_2)}{\partial x} &= 0 \\ \frac{\partial \alpha_1 \rho_1 e_1}{\partial t} + \frac{\partial \alpha_1 \rho_1 e_1 u}{\partial t} + \alpha_1 p_1 \frac{\partial u}{\partial x} &= \\ \frac{\partial \alpha_2 \rho_2 e_2}{\partial t} + \frac{\partial \alpha_2 \rho_2 e_2 u}{\partial t} + \alpha_2 p_2 \frac{\partial u}{\partial x} &= \end{aligned}$$

$$\begin{aligned}&\mu(p_1 - p_2) \\ &- p_I \mu(p_1 - p_2) \\ &p_I \mu(p_1 - p_2)\end{aligned}$$



$$\begin{aligned}\frac{\partial \alpha_1}{\partial t} &= \mu(p_1 - p_2) \\ \frac{\partial \alpha_1 \rho_1 e_1}{\partial t} &= -p_I \mu(p_1 - p_2) \\ \frac{\partial \alpha_2 \rho_2 e_2}{\partial t} &= p_I \mu(p_1 - p_2) \\ \frac{\partial \alpha_1 \rho_1}{\partial t} &= 0 \\ \frac{\partial \alpha_2 \rho_2}{\partial t} &= 0 \\ \frac{\partial \rho u}{\partial t} &= 0 \\ \frac{\partial \rho E}{\partial t} &= 0\end{aligned}$$

Relaxation system

2nd step: Pressure relaxation

$$\left. \begin{array}{l} \frac{\partial \alpha_1}{\partial t} = \mu(p_1 - p_2) \\ \frac{\partial \alpha_1 p_1 e_1}{\partial t} = -p_I \mu(p_1 - p_2) \\ \frac{\partial \alpha_2 p_2 e_2}{\partial t} = p_I \mu(p_1 - p_2) \end{array} \right\} \quad \begin{array}{l} \frac{de_1}{dt} + p_I \frac{dv_1}{dt} = 0 \\ \frac{de_2}{dt} + p_I \frac{dv_2}{dt} = 0 \end{array} \quad \text{time integration} \quad \rightarrow \quad \boxed{e_k - e_k^0 + \hat{p}_{Ik} (v_k - v_k^0) = 0}$$

$$\hat{p}_{Ik} = \frac{1}{v_k - v_k^0} \int_0^{\Delta t} p_I \frac{dv_k}{dt} dt = 0$$

$$\sum_k \Rightarrow Y_1 e_1 - Y_1 e_1^0 + Y_2 e_2 - Y_2 e_2^0 + \hat{p}_{II} (Y_1 v_1 - Y_1 v_1^0) + \hat{p}_{I2} (Y_2 v_2 - Y_2 v_2^0) = 0$$

$$\Leftrightarrow e - e^0 + (\hat{p}_{II} - \hat{p}_{I2}) (Y_1 v_1 - Y_1 v_1^0) = 0 \quad \rightarrow \quad \hat{p}_{II} = \hat{p}_{I2} = \hat{p}_I$$

Using mass equations

$$\text{Possible choice} \quad \hat{p}_I = p$$

Entropy inequality is verified

2nd step: Pressure relaxation

Using EOS :

$$\left. \begin{array}{l} e_1(p, v_1) - e_1^0(p_1^0, v_1^0) + p(v_1 - v_1^0) = 0 \\ e_2(p, v_2) - e_2^0(p_2^0, v_2^0) + p(v_2 - v_2^0) = 0 \end{array} \right\}$$
$$\left. \begin{array}{l} v_1 = v_1(p) \\ v_2 = v_2(p) \end{array} \right\}$$


Closure relation: $\alpha_1 + \alpha_2 = 1 \Leftrightarrow (\alpha\rho)_1^0 v_1(p) + (\alpha\rho)_2^0 v_2(p) = 1$



Zero function to solve

$$f(p) = (\alpha\rho)_1^0 v_1(p) + (\alpha\rho)_2^0 v_2(p) - 1$$

Then, we determine: $p \rightarrow v_k(p) \rightarrow \alpha_k = (\alpha\rho)_k v_k$

3th step: Internal energy reset

- We have in the 2nd step determined: p, v_k, α_k
- We forget the relaxed pressure but keep volume fractions:

$$\cancel{p \rightarrow \alpha_k}$$

- It is then possible to determine mixture pressure by the mixture EOS. By this way, energy conservation is ensured:

$$p_{\text{new}}(\rho, e, \alpha_1, \alpha_2) = \frac{\rho e - \left(\frac{\alpha_1 \gamma_1 p_{\infty 1}}{\gamma_1 - 1} + \frac{\alpha_2 \gamma_2 p_{\infty 2}}{\gamma_2 - 1} \right)}{\frac{\alpha_1}{\gamma_1 - 1} + \frac{\alpha_2}{\gamma_2 - 1}}$$

- Phasic EOS permits to reset internal energies:

$$e_k = e_k(p_{\text{new}}, \alpha_k \rho_k, \alpha_k)$$



Conservative and good treatment of wave dynamics from both sides of the interface

1D example: Water-Air shock tube

$$P_{\text{eau}} = 10\,000 \text{ bar}$$

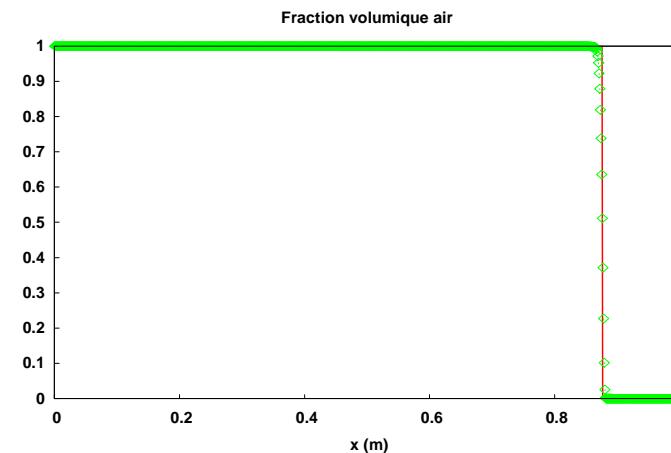
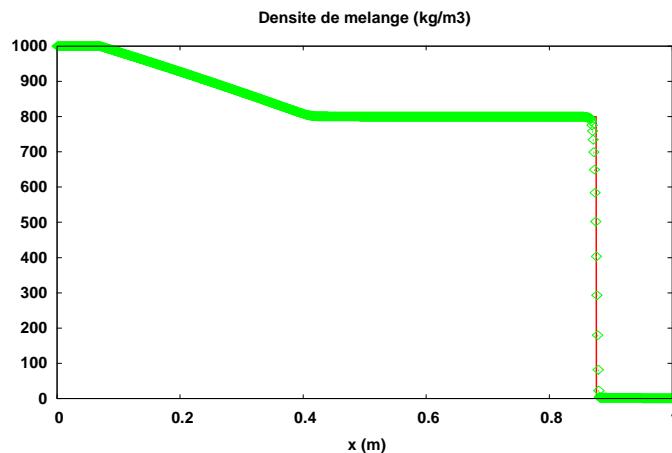
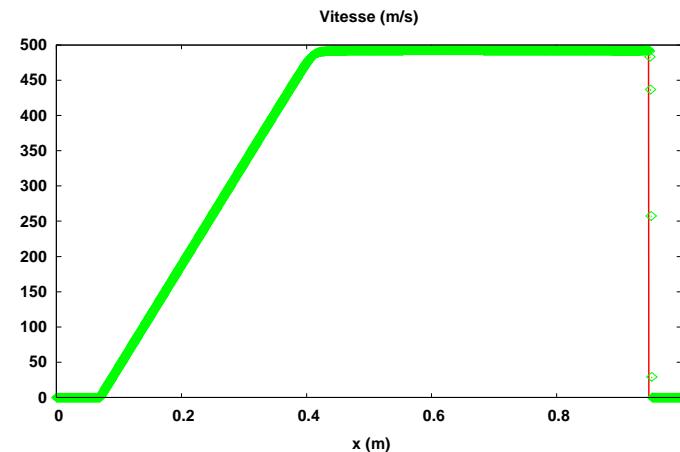
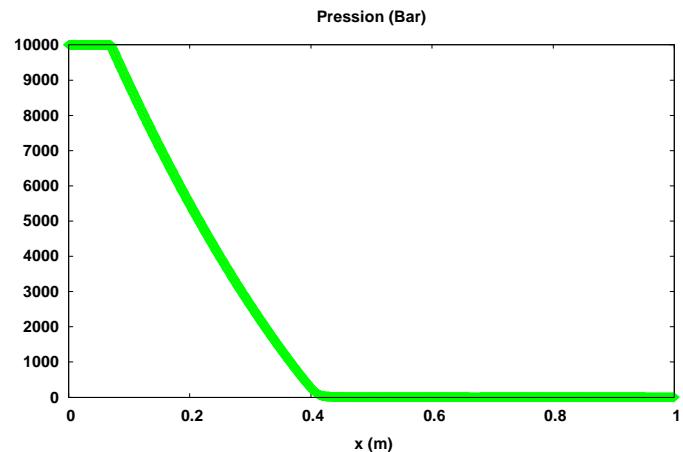
$$\rho_{\text{eau}} = 1000 \text{ kg/m}^3$$

EAU

$$P_{\text{eau}} = 1 \text{ bar}$$

$$\rho_{\text{eau}} = 1 \text{ kg/m}^3$$

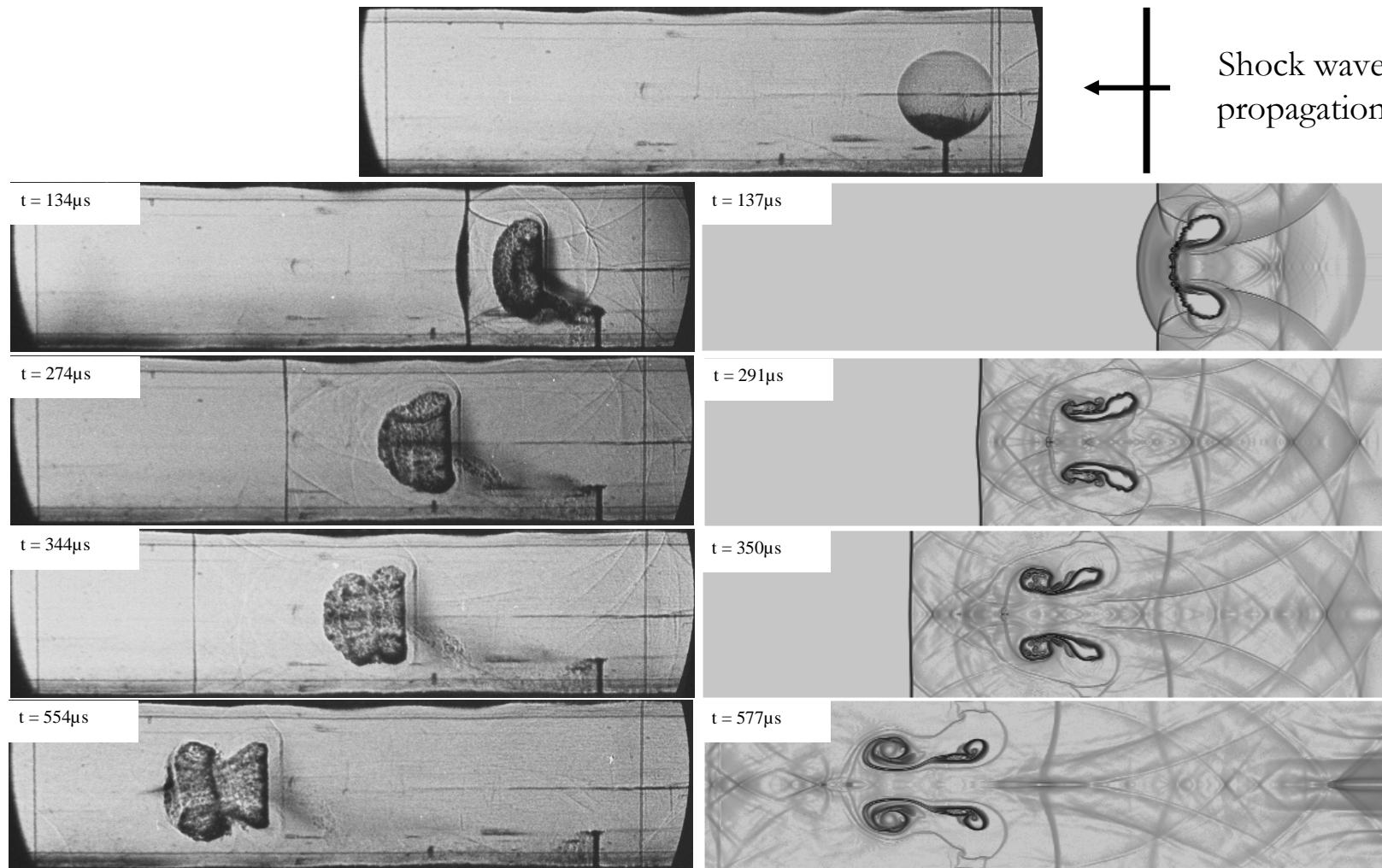
AIR



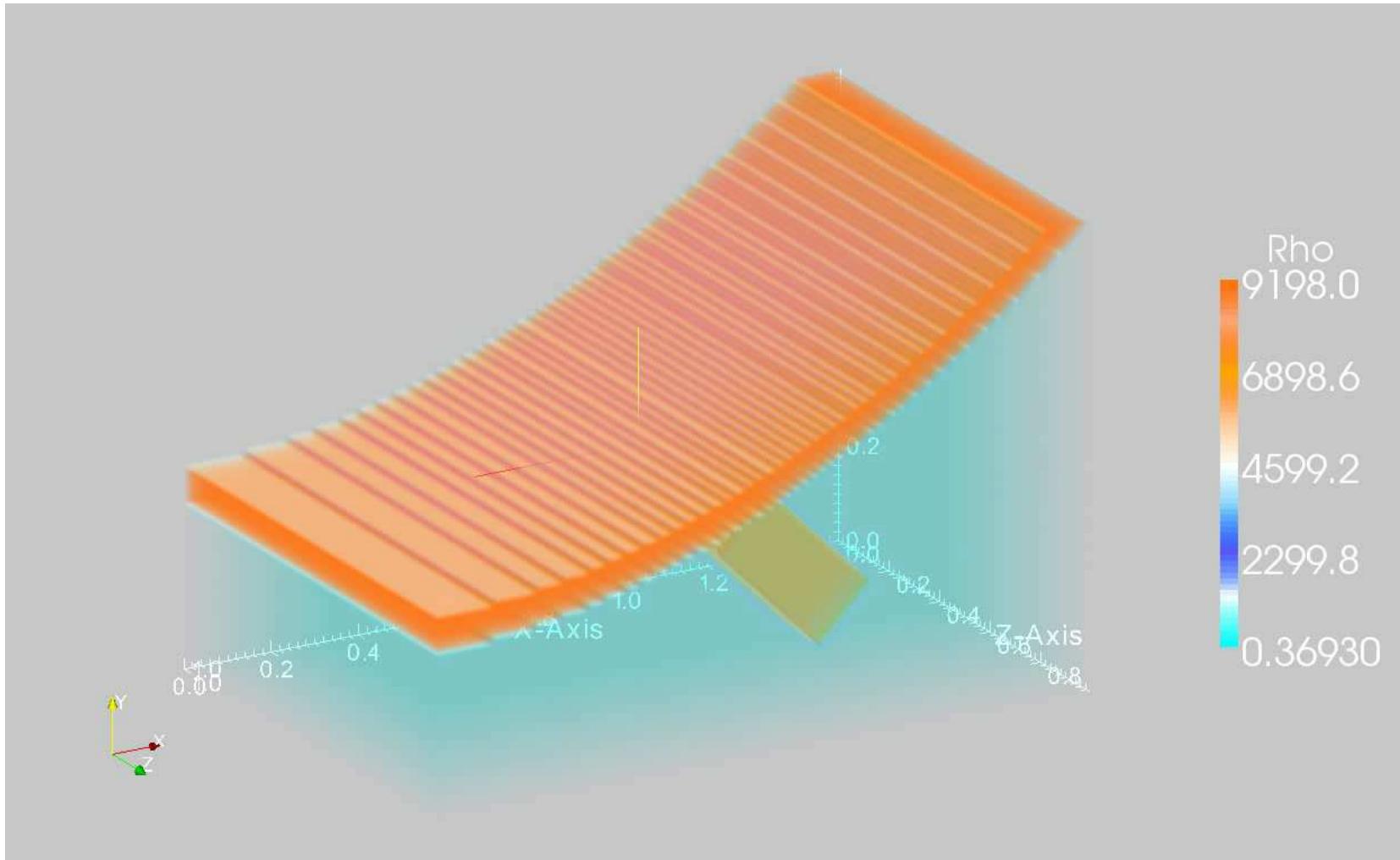
— Solution exacte

◇ Solution numérique

Résultats 2D - Expériences IUSTI (Layes, Jourdan, Houas)



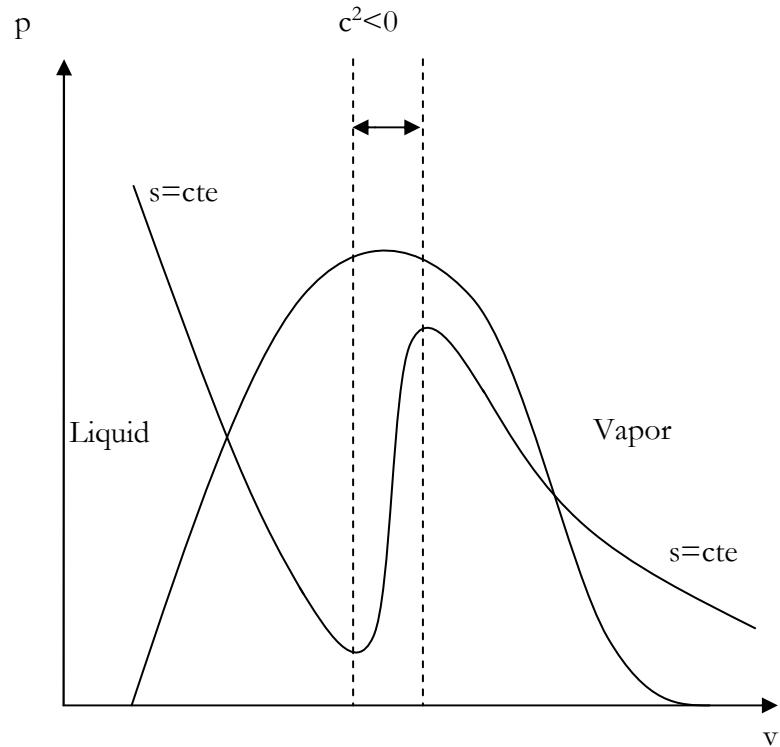
3D example : Missile impact on a metal plate





Topic 3 : Phase transition with the 5-equation model

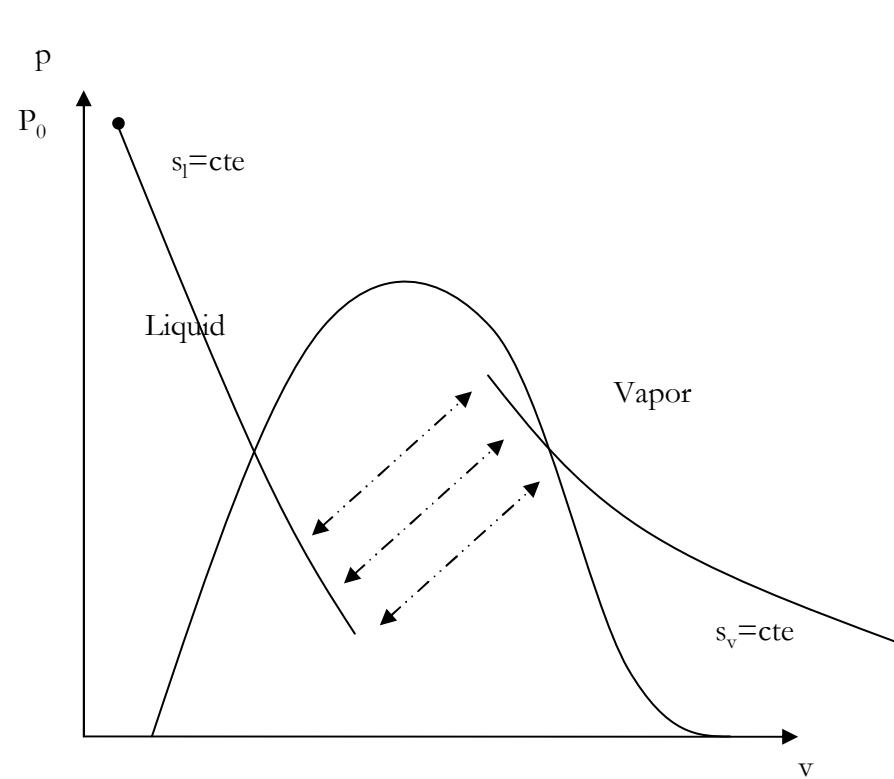
Why is the 5-equation model a good candidate for phase transition ?



Thermodynamic path using the Van de Waals representation

$$c^2 = -v^2 \frac{\partial P}{\partial v} \Bigg|_{s=cte} < 0$$

➡ Hyperbolicity is lost into the phase diagram



Kinetic path with the 5-equation model and two separate EOS

Liquid and Vapor isentropes are linked by a kinetic process

➡ Hyperbolicity is preserved in the entire domain

Phase transition modeling

1) Mass transfer modifies mass equations:

$$\frac{\partial \alpha_1 \rho_1}{\partial t} + \operatorname{div}(\alpha_1 \rho_1 \vec{u}) = \dot{m}_1 = \rho \dot{Y}_1$$

$$\frac{\partial \alpha_2 \rho_2}{\partial t} + \operatorname{div}(\alpha_2 \rho_2 \vec{u}) = -\dot{m}_1 = -\rho \dot{Y}_1$$

Avec $\dot{Y}_1 = \frac{dY_1}{dt} = \frac{d}{dt} \left(\frac{\alpha_1 \rho_1}{\rho} \right)$

2) Volume fractions change during phase transition:

$$\frac{d\alpha_1}{dt} = K \operatorname{div}(\vec{u}) + A Q_1 + \frac{\dot{m}_1}{\rho_I}$$

This « interfacial » density has to be determined in order to close the model.

Two-phase model for interface problems with phase transition

$$\begin{aligned}
 \frac{\partial \alpha_1 \rho_1}{\partial t} + \operatorname{div}(\alpha_1 \rho_1 \vec{u}) &= \rho v(g_2 - g_1) && \text{Entropy analysis for each phase and for the mixture} \\
 \frac{\partial \alpha_2 \rho_2}{\partial t} + \operatorname{div}(\alpha_2 \rho_2 \vec{u}) &= -\rho v(g_2 - g_1) \\
 \frac{\partial \rho \vec{u}}{\partial t} + \operatorname{div}(\rho \vec{u} \otimes \vec{u}) + \vec{\nabla}(p) &= 0 \\
 \frac{\partial \rho E}{\partial t} + \operatorname{div}((\rho E + p)\vec{u}) &= 0 \\
 \frac{\partial \alpha_1}{\partial t} + \vec{u} \cdot \vec{\nabla}(\alpha_1) &= \frac{(\rho_2 c_2^2 - \rho_1 c_1^2)}{\rho_1 c_1^2 + \rho_2 c_2^2} \operatorname{div}(\vec{u}) + \rho v(g_2 - g_1) \frac{\frac{c_1^2}{\alpha_1} + \frac{c_2^2}{\alpha_2}}{\frac{\rho_1 c_1^2}{\alpha_1} + \frac{\rho_2 c_2^2}{\alpha_2}} + H(T_2 - T_1) \frac{\frac{\Gamma_1}{\alpha_1} + \frac{\Gamma_2}{\alpha_2}}{\frac{\rho_1 c_1^2}{\alpha_1} + \frac{\rho_2 c_2^2}{\alpha_2}}
 \end{aligned}$$

Mechanical relaxation Mass transfer Heat transfer

Lois d'état :

$$\rho e = \alpha_1 \rho_1 e_1 + \alpha_2 \rho_2 e_2$$

$$\left\{
 \begin{array}{l}
 e_1(\rho_1, p) = \frac{p + \gamma_1 p_{\infty,1}}{(\gamma_1 - 1) \rho_1} + e_{0,1} \quad \text{For the liquid} \\
 e_2(\rho_2, p) = \frac{p}{(\gamma_2 - 1) \rho_2} + e_{0,2} \quad \text{For vapor}
 \end{array}
 \right.$$

$$p(\rho, e, \alpha_1, \alpha_2) = \frac{\rho e - \left(\frac{\alpha_1 \gamma_1 p_{\infty,1}}{\gamma_1 - 1} + \frac{\alpha_2 \gamma_2 p_{\infty,2}}{\gamma_2 - 1} \right)}{\frac{\alpha_1}{\gamma_1 - 1} + \frac{\alpha_2}{\gamma_2 - 1}}$$

Kinetic parameters : $v, H = \begin{cases} +\infty & \text{at interfaces (thermodynamical equilibrium)} \\ 0 & \text{elsewhere (metastable state)} \end{cases}$

Thermodynamic closure

Assumption : Each fluid is governed by the stiffened gas EOS:

$$p(\rho, e) = (\gamma - 1) \underbrace{\rho(e - q)}_{\text{Repulsive effects (gas, liquids and solids)}} - \gamma p_\infty \underbrace{-}_{\text{Attractive effects (liquids and solids)}}$$

Repulsive effects (gas, liquids and solids)

Attractive effects (liquids and solids)

Other useful forms for the SG EOS :

$$h(T) = \gamma C_v T + q$$

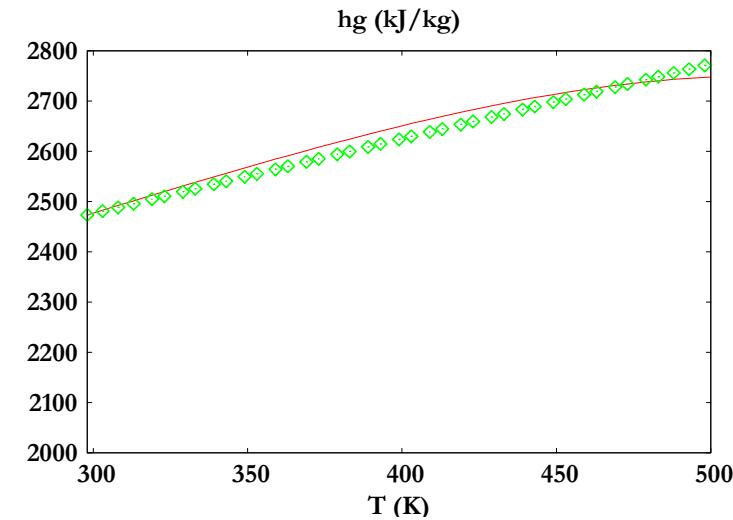
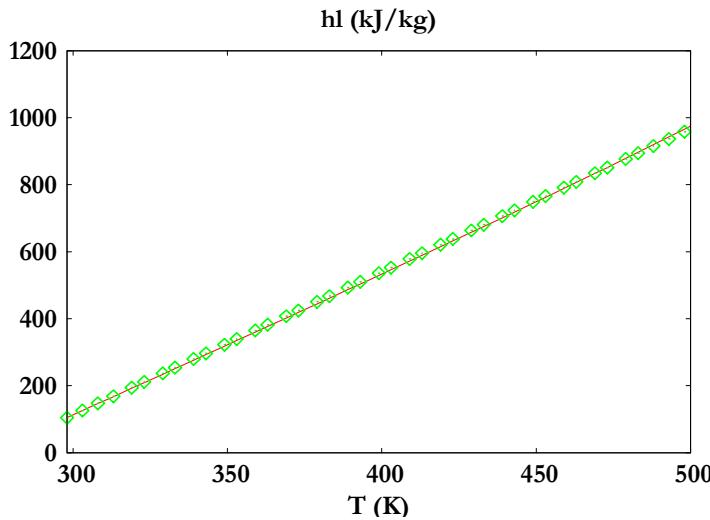
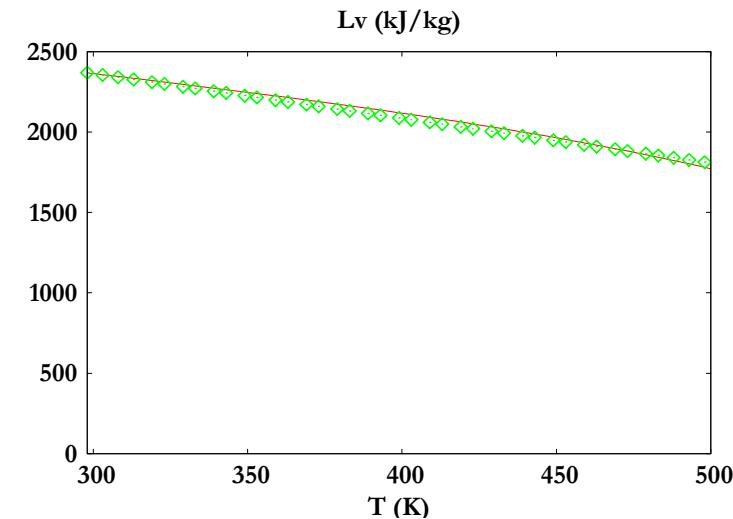
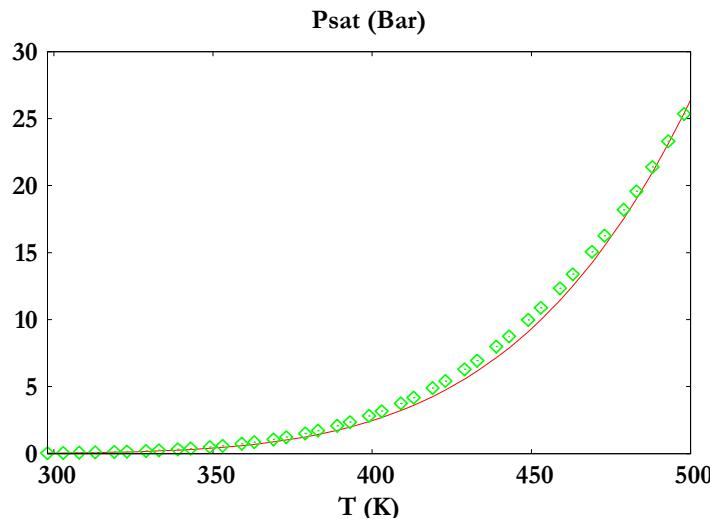
$$s(p, T) = C_v \ln \frac{T^\gamma}{(p + p_\infty)^{\gamma-1}} + q'$$

$$g(p, T) = (\gamma C_v - q')T - C_v T \ln \frac{T^\gamma}{(p + p_\infty)^{\gamma-1}} + q$$

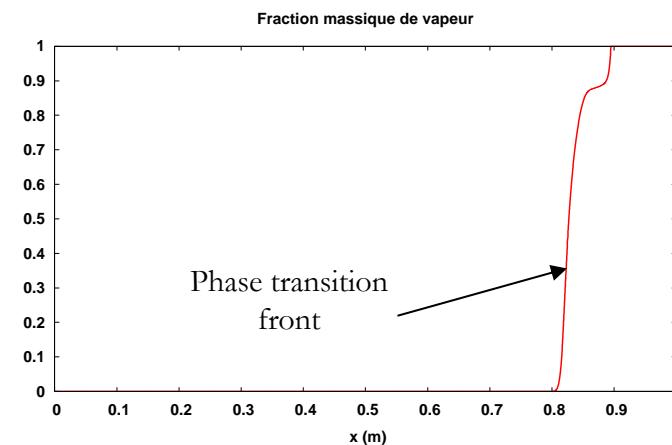
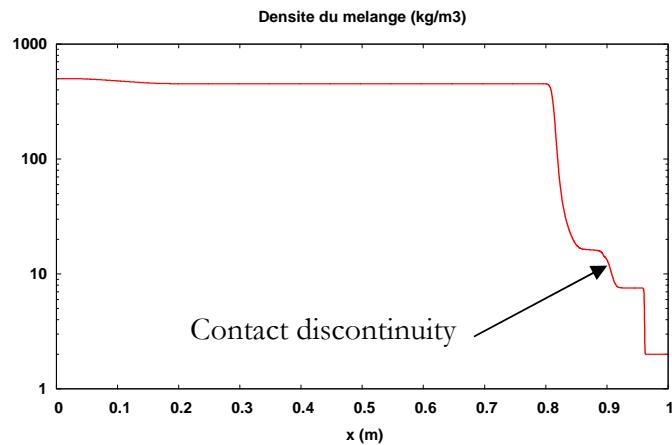
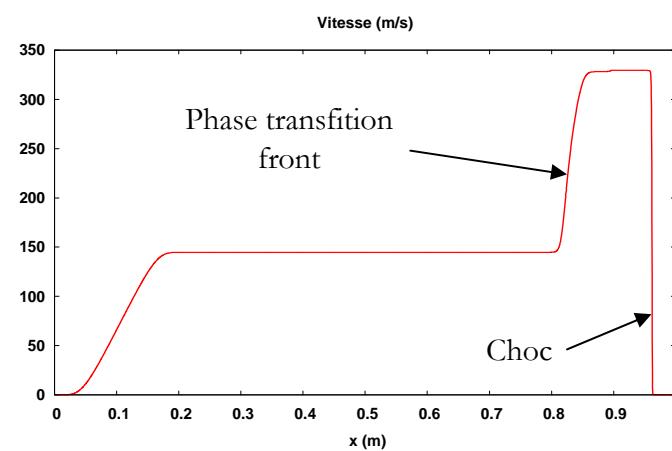
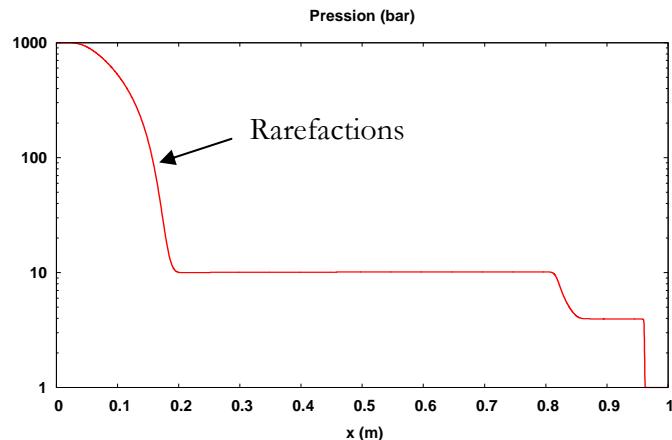
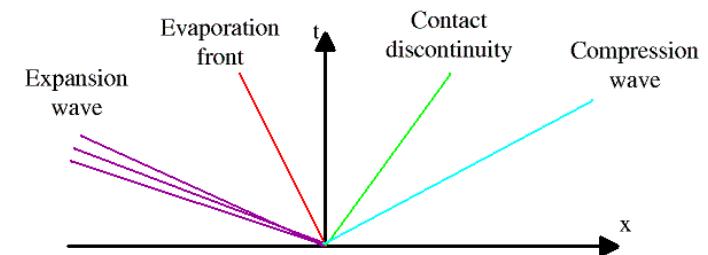


For each fluid EOS : 5 parameters to determine : $\gamma, P_\infty, C_v, q, q'$

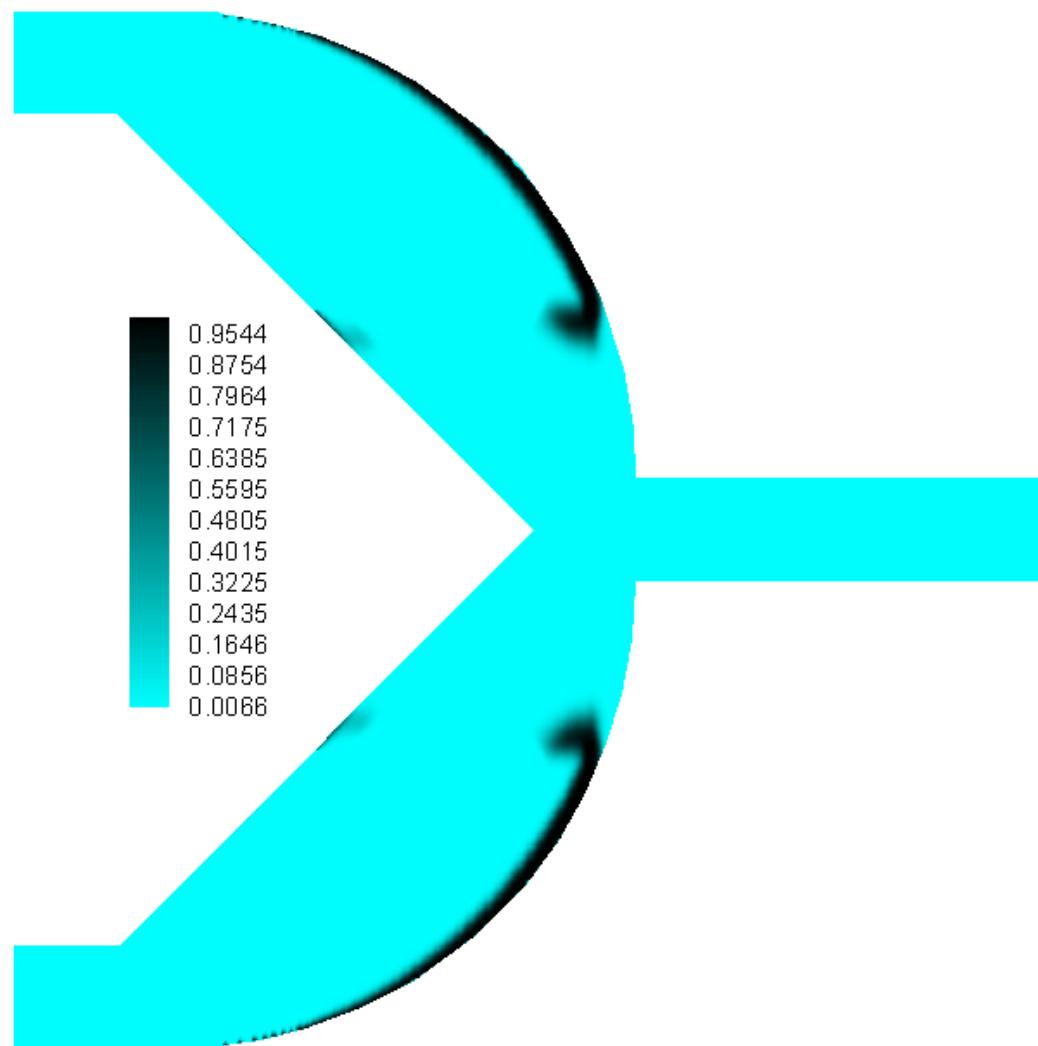
Saturation curves for liquid water and vapor water



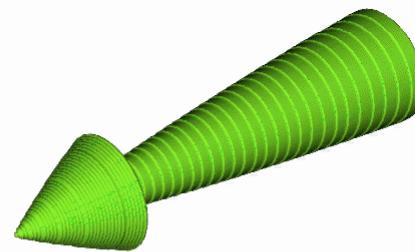
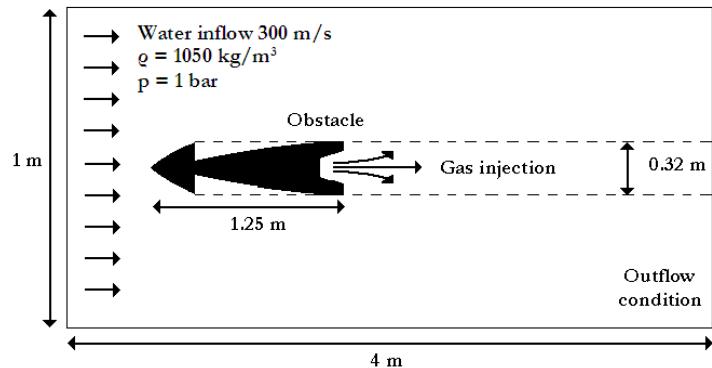
Dodécane liquide	Dodécane vapeur
$P_{\text{liquide}} = 1000 \text{ bar}$ $\rho_{\text{liquide}} = 500 \text{ kg/m}^3$	$P_{\text{vapeur}} = 1 \text{ bar}$ $\rho_{\text{vapeur}} = 2 \text{ kg/m}^3$



High pressure diesel injector



High speed motion under water

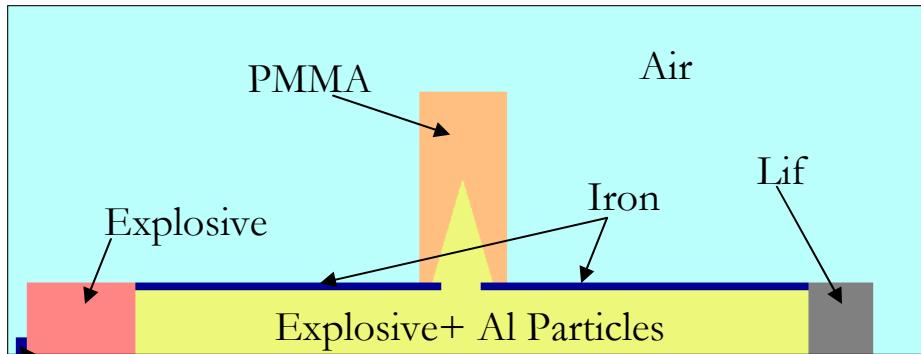


Topic 4 : Some possible extensions

- Capillary effects (Perigaud & Saurel, 2005)
- Detonation (Petitpas et al. 2009)
- Gravity, heat conduction, viscosity, turbulence, etc.

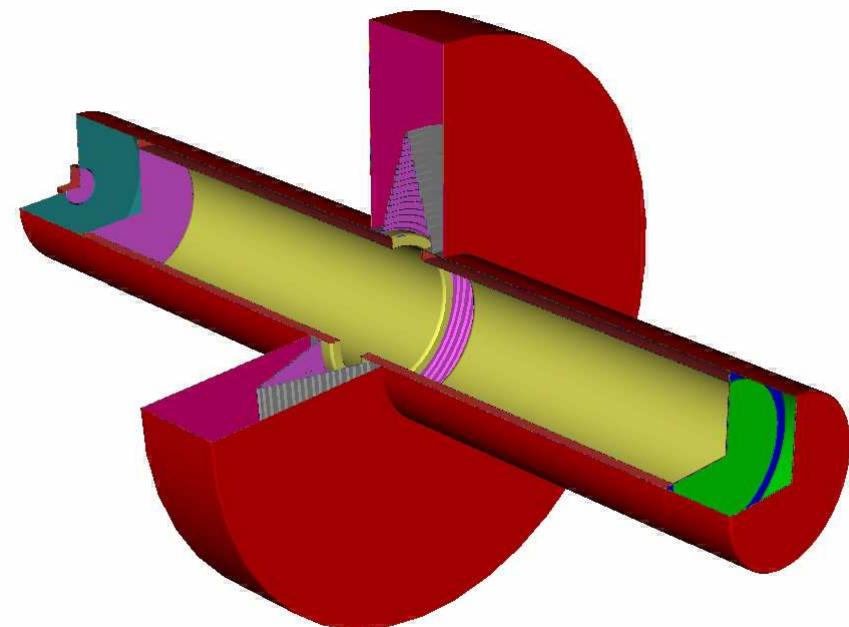
Detonation problems

Russian experiences done for the DGA



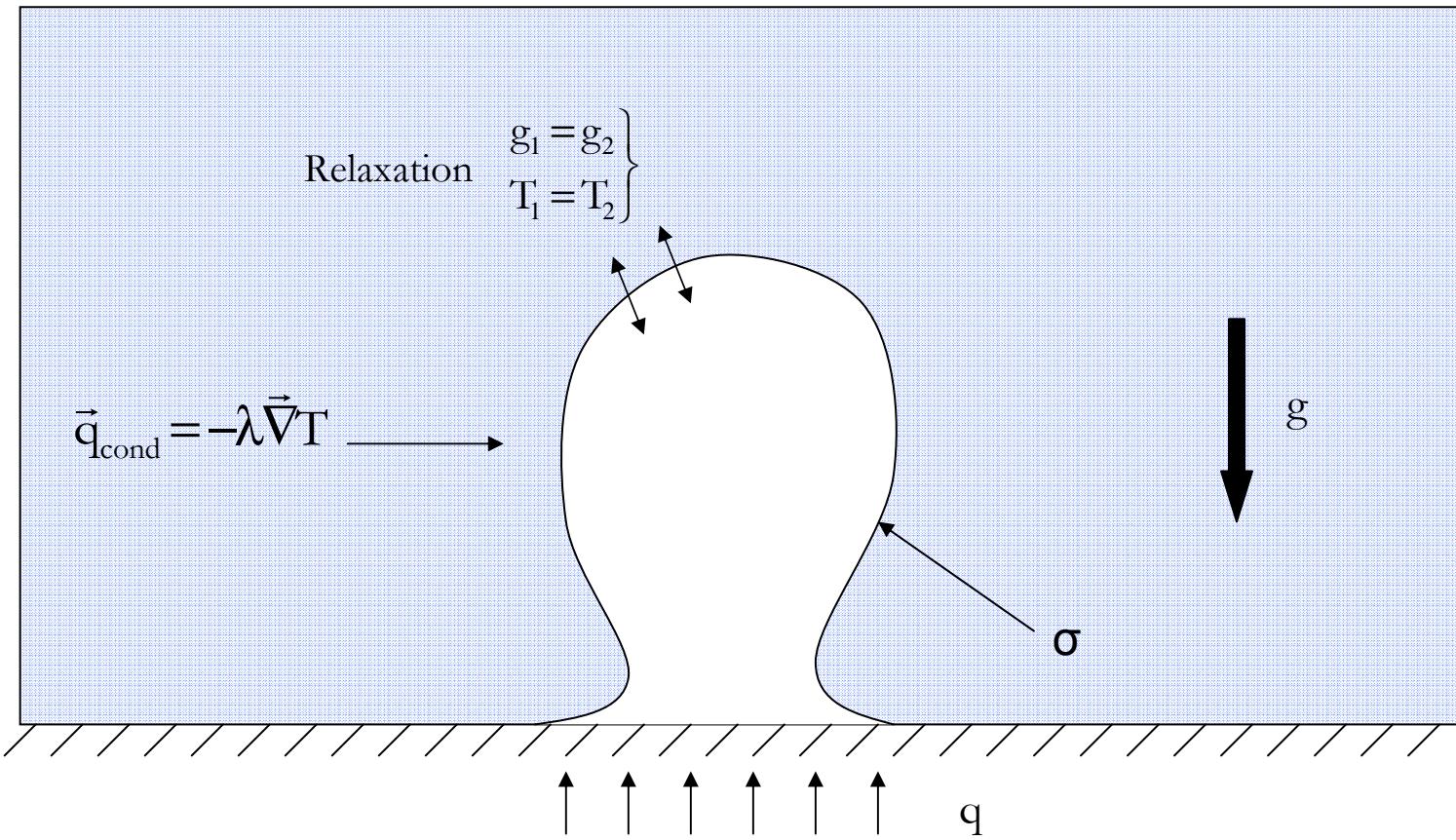
High velocity
impact for ignition
(1500 m/s)

- 7 fluids
- EOS SG, JWL, IG
- Density ratio : 9000



Ebullition crisis simulation

Many physical ingredients are required



Bubble growth

